

UNCLASSIFIED

AD NUMBER	
AD382651	
CLASSIFICATION CHANGES	
TO:	UNCLASSIFIED
FROM:	CONFIDENTIAL
LIMITATION CHANGES	
TO: Approved for public release; distribution is unlimited.	
FROM: Distribution authorized to U.S. Gov't. agencies and their contractors; Critical Technology; JUN 1967. Other requests shall be referred to Air Force Rocket Propulsion Laboratory, RPPR-STINFO, Edwards AFB, CA 93523. This document contains export-controlled technical data.	
AUTHORITY	
AFRPL ltr, 18 Jan 1978 AFRPL ltr, 18 Jan 1978	

THIS PAGE IS UNCLASSIFIED

GENERAL DECLASSIFICATION SCHEDULE

**IN ACCORDANCE WITH
OOD 5200.1-R & EXECUTIVE ORDER 11652**

THIS DOCUMENT IS:

CLASSIFIED BY _____

**Subject to General Declassification Schedule of
Executive Order 11652-Automatically Downgraded at
2 Years Intervals- DECLASSIFIED ON DECEMBER 31,**

BY

**Defense Documentation Center
Defense Supply Agency
Cameron Station
Alexandria, Virginia 22314**

THIS REPORT HAS BEEN DELIMITED
AND CLEARED FOR PUBLIC RELEASE
UNDER DOD DIRECTIVE 5200.20 AND
NO RESTRICTIONS ARE IMPOSED UPON
ITS USE AND DISCLOSURE.

DISTRIBUTION STATEMENT A

APPROVED FOR PUBLIC RELEASE;
DISTRIBUTION UNLIMITED.

SECURITY

MARKING

The classified or limited status of this report applies to each page, unless otherwise marked.

Separate page printouts MUST be marked accordingly.

THIS DOCUMENT CONTAINS INFORMATION AFFECTING THE NATIONAL DEFENSE OF THE UNITED STATES WITHIN THE MEANING OF THE ESPIONAGE LAWS, TITLE 18, U.S.C., SECTIONS 793 AND 794. THE TRANSMISSION OR THE REVELATION OF ITS CONTENTS IN ANY MANNER TO AN UNAUTHORIZED PERSON IS PROHIBITED BY LAW.

NOTICE: When government or other drawings, specifications or other data are used for any purpose other than in connection with a definitely related government procurement operation, the U. S. Government thereby incurs no responsibility, nor any obligation whatsoever; and the fact that the Government may have formulated, furnished, or in any way supplied the said drawings, specifications, or other data is not to be regarded by implication or otherwise as in any manner licensing the holder or any other person or corporation, or conveying any rights or permission to manufacture, use or sell any patented invention that may in any way be related thereto.

CONFIDENTIAL

AFRPL-TR-67 -165

(UNCLASSIFIED)

**COMBUSTION MECHANISM
OF
HIGH BURNING RATE SOLID PROPELLANTS**

Contract F04611-67-C-0034

**QUARTERLY TECHNICAL REPORT AFRPL-TR-67 -165
June 1967**

**David A. Flanigan
Huntsville Division
Thiokol Chemical Corporation**

In addition to security requirements which must be met, this document is subject to special export controls and each transmittal to foreign governments or foreign nationals may be made only with prior approval of AFRPL (RPPR-STINFO), Edwards, California 93523

DOWNGRADING INFORMATION

Downgraded at 3 year intervals. Declassified after 12 years.
DOD Dir 5200.10.

DEFENSE INFORMATION

This document contains information affecting the national defense of the United States within the meaning of the Espionage Laws, Title 18 U.S.C., Sections 793 and 794. The transmission or the revelation of its contents in any manner to an unauthorized person is prohibited by law.

**Air Force Rocket Propulsion Laboratory
Research and Technical Division
Air Force Systems Command
United States Air Force
Edwards Air Force Base, California 93523**

CONFIDENTIAL

AD382651

NOTICES

When U. S. Government drawings, specifications, or other data, are used for any purpose other than a definitely related Government procurement operation, the Government thereby incurs no responsibility nor any obligation whatsoever; and the fact that the Government may have formulated, furnished, or in any way supplied the said drawings, specifications, or other data, is not to be regarded by implication or otherwise, as in any manner licensing the holder or any other person or corporation, or conveying any rights or permission to manufacture, use or sell any patented invention that may be related thereto in any way.

Secrecy Order Notice, U. S. Patent Office as Modified by Permit A:
This document contains information which is subject to secrecy orders issued by the U. S. Commissioner of Patents in pursuance of Title 35, United States Code (1952), Sections 181-188.

CONFIDENTIAL

(UNCLASSIFIED)
COMBUSTION MECHANISM
OF
HIGH BURNING RATE SOLID PROPELLANTS

Contract F04611-67-C-0034

QUARTERLY TECHNICAL REPORT AFRPL-TR-67-165

David A. Flanigan

DOWNGRADING INFORMATION

Downgraded at 5 year intervals, Declassified after 12 years, DOD Dir 5200.10.

DEFENSE INFORMATION

This document contains information affecting the national defense of the United States within the meaning of the Espionage Laws, Title 18 U.S.C., sections 793 and 794. The transmission or the revelation of its contents in any manner to an unauthorized person is prohibited by law.

In addition to security requirements which must be met, this document is subject to special export controls and such transmittal to foreign governments or foreign nationals may be made only with prior approval of AFRPL (RPPR-STINFO), Edwards, California 93523.

CONFIDENTIAL

CONFIDENTIAL

(This page is UNCLASSIFIED)

FOREWORD

(U) This, the second Quarterly Technical Report under Contract No. F04611-67-C-0034, covers the work performed from 1 March through 31 May 1967. This contract with the Huntsville Division of Thiokol Chemical Corporation was initiated under Air Force Rocket Propulsion Laboratory, Research and Technology Division Project Number 3148. It is being accomplished under the technical direction of R. W. Bargmeyer, 1/Lt., USAF of the Research and Technology Division, Air Force Systems Command, United States Air Force, Edwards Air Force Base, California 93523.

(U) Dr. David A. Flanigan of Thiokol's Research and Development Department is the Principal Investigator and Mr. Carl J. Whelchel of the Project Management Directorate is Assistant Project Manager for this program. Full authority for the management control of this program is the responsibility of Mr. G. F. Mangum of the Project Management Directorate. Others who cooperated in the work and in the preparation of this report are Messrs. B. A. Allen, C. S. Combs, C. I. Ashmore and Mrs. E. J. Grice.

(U) This report has been assigned the Thiokol internal number 39-67 (Control No. C-67-39A).

(U) This project is being accomplished as a part of the Air Force program, the overall objective of which is to tailor the burning rate of a propellant predictably and controllably to any desired level in the range from 1 to 10 inches per second. Experiments will be performed to synthesize more efficient burn rate catalysts by maximizing guideline properties determined under Contract AF04(611)-11212 toward development of an ideal catalyst. Following catalyst synthesis, each compound showing potential will be subjected to comprehensive decomposition studies and combustion mechanism evaluation with propellant ingredients.

(U) This report contains no classified information extracted from other classified documents.

STATEMENT OF APPROVAL

(U) Publication of this report does not constitute Air Force approval of the report's findings or conclusions. It is published only for the exchange and stimulation of ideas.

W. H. EBELKE, Col., USAF
Chief, Propellants Division
Air Force Rocket Propulsion Laboratory

CONFIDENTIAL

CONFIDENTIAL

CONFIDENTIAL ABSTRACT

(C) Thiokol's program to tailor the burning rate of a propellant predictably and controllably to any desired level in the range from 1 to 10 inches per second is divided into three phases: Phase I - Synthesis of Burning Rate Catalysts, Phase II - Decomposition Studies and Evaluation of Catalysts and Phase III - Decomposition of Advanced Oxidizers, Fuels, and Binders. Experiments will be performed to synthesize more efficient burn rate catalysts by maximizing already determined guideline properties toward development of an ideal catalyst under Phases I and II. Following catalyst synthesis, each compound showing potential will be subjected to comprehensive decomposition studies and combustion mechanism evaluation with propellant ingredients. To date, 30 candidate catalyst materials have been synthesized and the physical properties ascertained. Compatibility testing of these materials is essentially complete and no serious problems are expected in the formulation and scale-up of propellants containing the candidate catalysts. Burn rate studies thus far indicate that several prospective liquid ferrocene derivatives offer considerable potential in increasing burn rates over that presently available with n-butyl ferrocene [PLASTISCAT -IV^(R)].

CONFIDENTIAL

CONFIDENTIAL

(This page is UNCLASSIFIED)

TABLE OF CONTENTS

	<u>Page</u>
SECTION I	
INTRODUCTION	1
SECTION II	
EXPERIMENTAL ACCOMPLISHMENTS	3
1. Phase I - Synthesis of Burning Rate Catalysts	3
2. Phase II - Decomposition Studies and Catalyst Evaluation	10
a. Catalyst Compatibility Studies	10
b. Burn Rate Studies	31
SECTION III	
CONCLUSIONS	43

LIST OF TABLES

I	Physical Constants of Catalysts Evaluated	4
II	Catalyst Compatibility Thin-Layer Chromatograph Results (160° F)	26
III	Impact and Friction Sensitivity of Catalyst/Ammonium Perchlorate (5%)	27
IV	Physical Property Characteristics and Sensitivity Test Results of Catalyzed Propellant	37
V	Characteristics of Catalyzed Propellant	38

CONFIDENTIAL

(This page is UNCLASSIFIED)

LIST OF FIGURES

<u>Number</u>		<u>Page</u>
1.	DTA of Standard Grind AP (50-55 Micron W. M. D.) Control After 10 days at 160° F.	12
2.	DTA of AP (Standard Grind) + 5 Percent Catalyst Methoxymethyl Ferrocene	13
3.	DTA of AP (Standard Grind) + 5 Percent Catalyst Copper (I) Complex of β -Ferrocenyl- β -oxo-Propionaldehyde	14
4.	DTA of Copper (II) Complex of β -Ferrocenyl- β -oxo- Propionaldehyde	15
5.	DTA of 2-Methoxyethyl Ferrocenoate	16
6.	DTA of Copper Complex of 1, 3-diferrocenyl-1, 3- Propanedione	17
7.	DTA of 1, 3-Diferrocenyl-1, 3-Propanedione	18
8.	DTA of Dimethylaminomethyl Ferrocene	19
9.	DTA of Allyl Ferrocenoate	20
10.	DTA of n-Propyl Ferrocenoate	21
11.	DTA of 1, 3-Diferrocenyl-oxo-2-Propene	22
12.	DTA of Propargyl Ferrocenoate	23
13.	DTA of Mixture of 1, 1'-di (Methoxymethyl) Ferrocene and 1 hydroxymethyl-1'-Methoxymethyl Ferrocene	24
14.	Differential Thermal Analyses of Methyl- β -(ferrocenyl- methylthio) propionate, Ferrocenylmethyl ferrocenoate, and β -(ferrocenylmethylthio)ethanol.	28
15.	Differential Thermal Analysis of Ammonium Perchlorate (ung) + 5 % Catalyst.	29
16.	DTA of 5 Percent Catalyst with AP	30
17.	Percent Weight Loss versus Temperature by TGA Method - Copper (II) Complex β -ferrocenyl - β -oxo - propion- aldehyde	32
18.	Percent Weight Loss versus Temperature by TGA Method - dimethylaminomethyl ferrocene	33
19.	Percent Weight Loss versus Temperature by TGA Method - 1, 3-diferrocenyl-1 oxo-2-propene	34
20.	Comparison of the Burning Rate Characteristics of Propellants Catalyzed with PLASTISCAT-IV ^(R) and Methoxymethyl Ferrocene	35
21.	Comparison of the Burning Rate Characteristics of Propellants Catalyzed with 1, 3-diferrocenyl-1-oxo-2 propene, bis (α -ferrocenylethyl) ether, allyloxymethyl ferrocene and ethoxymethyl ferrocene	39
22.	Burning Rate Characteristics of Propellants Catalyzed with Ethyl Ferrocenylacetate and Methyl- β -(ferrocenylmethylthio) propionate.	41
23.	Burning Rate Characteristics of Propellants Catalyzed with Methyl Ferrocenylacetate and Mercaptomethyl Terminated poly-[bis (methylthiomethyl) ferrocene].	42

CONFIDENTIAL

SECTION I

INTRODUCTION

(U) The objective of this program is to tailor the burning rate of a propellant predictably and controllably, to any desired level in the range from 1 to 10 inches per second. Ballistic and mechanical properties of propellants studied will be maintained at the state-of-the-art standards of current Minuteman propellant.

(C) Thiokol's approach to obtain the program objective will be through the continued investigation of iron compound effects on the aluminum-ammonium perchlorate-polybutadiene binder system. It is also designed to systematically evaluate new propellant ingredients as to their effect on burning rate and the related effect on combustion mechanism. New materials which will be evaluated are hydroxyl-ammonium perchlorate, hydrazine diperchlorate and nitronium perchlorate oxidizers; aluminum hydride, beryllium, beryllium hydride fuels; P-BEP, NFPA polymers, and TVOPA plasticizer. The data and conclusions reached under Contract AF04(611)-11212 will be used as a base line guide for the work to be accomplished under this program.

(U) The planned program consists of three major areas: synthesis, evaluation, and advanced ingredient studies. It is designed to progress in a logical manner so that the most promising materials receive the more extensive evaluation and the less desirable materials are discarded at an early date. The three phases of the program are:

(U) Phase I - Synthesis of Burning Rate Catalysts

(U) Burning rate catalysts will be synthesized for use in the AP/Al/PB propellant system. Prior knowledge as a propulsion contractor and the data obtained from Contract AF04(611)-11212 will serve as base line guides. As a new material is synthesized, it will be evaluated in Phase II.

(U) Phase II - Decomposition Studies and Evaluation of Catalysts

(U) An evaluation of the compatibility of the candidate catalysts with other propellant ingredients will be accomplished upon completion of synthesis, elemental characterization, and physical property determination of each candidate material. Data obtained in this phase will allow recommendations to be made relative to the development of new burning rate catalysts and the development of high burning rate propellants utilizing the improved catalyst.

CONFIDENTIAL

CONFIDENTIAL

(This page is UNCLASSIFIED)

(U) Phase III - Decomposition of Advanced Oxidizers, Fuels, and Binders

(U) Phase III effort will be directed toward obtaining a basic fundamental knowledge of the decomposition of advanced fuels, oxidizers, and binders. Laboratory test data will be utilized to postulate a burning mechanism of the advanced ingredient and a comparison made with that of conventional propellant ingredients.

(U) This report covers work performed for the period 1 March through 31 May 1967 under Contract F04611-67-C-0034. Effort to date has been concerned solely with Phases I and II, which are being conducted concurrently. Phase III will be initiated after the completion of Phases I and II.

CONFIDENTIAL

SECTION II

EXPERIMENTAL ACCOMPLISHMENTS

(U) Based on the results of effort expended under Contract AF04(611)-11212, the experiments under this program will be directed toward the synthesis and characterization of more efficient burn rate catalysts.

(U) 1. Phase I - Synthesis of Burning Rate Catalysts

(U) The guidelines for synthesis of more efficient burn rate catalysts have been derived from effort conducted under the above contract and are listed below:

- High iron content
- Wide liquid range
- Readily oxidizable
- Compatible with other propellant ingredients
- Maximum fuel content (heat release on oxidation)

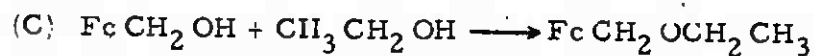
(U) The synthesis of newer burn rate catalyst will be directed toward maximizing the aforementioned properties into the ideal catalyst. Effort performed in this area of research during this reporting period is presented in subsequent paragraphs.

(U) Preliminary synthesis effort has afforded a series of both liquid and solid ferrocene containing compounds which have been evaluated as candidate burn rate catalysts. A list of the compounds as well as their physical constants is shown in Table I.

(U) During this report period, further synthesis effort has been accomplished with the following results:

(C) The reaction of hydroxymethyl ferrocene with ethanol in the presence of an acid catalyst gave ethoxymethyl ferrocene (b. p. $88^{\circ}\text{C}/0.15\text{ mm}$).

Reaction:



CONFIDENTIAL

TABLE I
PHYSICAL CONSTANTS OF CATALYSTS EVALUATED

Compound	M.p., °C	B.p., °C	Theo. %Fe	Theo. %Cu
Dimethylaminomethyl ferrocene	---	78/0.02 mm	23	---
Propargyl ferrocenoate	81-84 (dec.)	---	21	---
Allyl ferrocenoate	36-36.5	122/0.15 mm	21	---
n-Propyl ferrocenoate	---	96/0.10 mm	21	---
1,3-Diferrocenyl-1-oxo-2-propene	198-199	---	26	---
Sodium salt of β -ferrocenyl- β -oxo propionaldehyde	240-242 (dec.)	---	20	---
1,3-Diferrocenyl-1,3-propanedione	215 (dec.)	---	25	---
Copper complex of 1,3-diferrocenyl-1,3-propanedione	280 (dec.)	---	24	7
2-methoxyethyl ferrocenoate	34-35	---	19	---
Copper (I) complex of β -ferrocenyl- β -oxopropionaldehyde	224-225 (dec.)	---	18	20
Copper (II) complex of β -ferrocenyl- β -oxopropionaldehyde	200 (dec.)	---	20	11
Methoxy methyl ferrocene	---	66/0.025	24	---
1,1'-Di(methoxy methyl) ferrocene	---	310	20	---
1-Hydroxy methyl-1'-methoxy methyl ferrocene	---	280	22	---
Trimethoxy iron	> 300	---	38	---

CONFIDENTIAL

CONFIDENTIAL

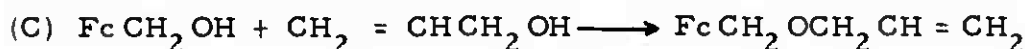
(U) Bisferrocenylmethylether (m.p. 126 - 129° C; Lit¹ 130 - 131° C) was prepared by the addition of acid catalyst to an ethereal solution of hydroxymethyl ferrocene.

Reaction:



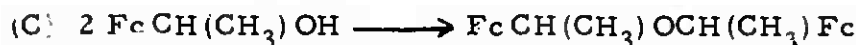
(U) Allyloxymethyl ferrocene [b.p., 260° C (DTA)] was prepared by the reaction of hydroxymethyl ferrocene with allyl alcohol in the presence of acid.

Reaction:



(U) Bis (α - ferrocenylethyl) ether [b.p. 197° C, 232° C (dec.) (DTA)] was prepared by the reaction of α - hydroxyethyl ferrocene with an acid catalyst.

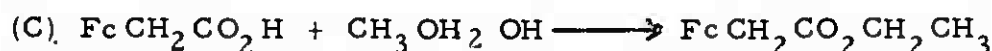
Reaction:



(C) Attempts to prepare triferrocenylmethoxy iron by the reaction of hydroxymethyl ferrocene with ferric chloride gave bisferrocenylmethyl ether (ca. 3 percent yield) and an equal amount of a brown solid, which melted at 99 - 101° C. The infrared spectrum did not rule out triferrocenylmethoxy iron, but the difficulty involved in isolation of this compound and the low yield obtained discourage further investigation of this compound as a serious candidate.

(C) Esterification of ferrocenylacetic acid with methanol gave methyl ferrocenylacetate [b.p. 282° C (DTA)].

Reaction:

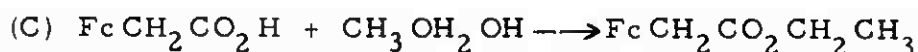


-
1. E. G. P. Nevalova, Yu. A. Ustynyuk, and A. N. Nesmeyanov, Izv. Akad. Nauk SSSR. Ser. Khim., No. 11, 1972 (1963), (English Translation). Also see C. R. Hauser and C. E. Cain, J. Org. Chem., 23, 2007 (1958).

CONFIDENTIAL

Substitution of ethanol in this preparation afforded ethyl ferrocenylacetate [b. p. 286°C (DTA)].

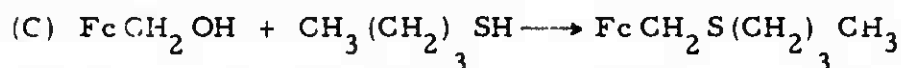
Reaction:



(C) Attempted purification of the products using base-washed alumina was not successful as they turned dark brown on the column. Acid-washed alumina, however, has been successfully employed with no apparent ill effects.

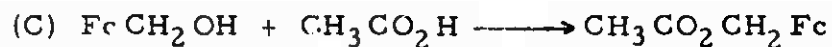
(C) The reaction of hydroxymethyl ferrocene with butyl mercaptan in the presence of an acid catalyst gave n-butylthiomethyl ferrocene [b. p. 274°C , 300°C (dec.) (DTA)].

Reaction:

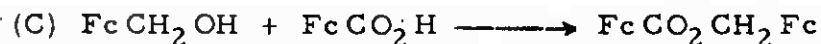


(C) The reaction of 1,1'-dihydroxymethyl ferrocene with methanol and water (acetic acid catalyst) gives a mixture of 1,1'-dimethoxymethyl ferrocene and 1-hydroxymethyl-1'-methoxymethyl ferrocene (previously reported). This material has been subjected to low temperatures for extended periods ($+10^{\circ}\text{C}$ for 24 hours; -10°C for 72 hours; -40°C for 48 hours) and has, thus far, failed to freeze. It should be noted that the material was exposed to the above temperatures in the order given and the cooling was, therefore, fairly gradual.

(C) Reaction of hydroxymethyl ferrocene with acetic acid gave ferrocenylmethyl acetate (m. p., 75 to 77°C) in good yield.



(C) The reaction of hydroxymethyl ferrocene with ferrocenoic acid in acetone afforded ferrocenylmethyl ferrocenoate (m. p. 131 to 135°C).



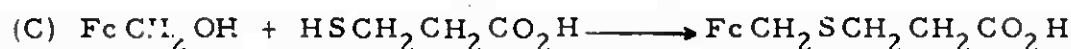
(C) An attempt was made to prepare ferrocenylacetic acid by reaction of acetylferrocene with sodamide and subsequent carboxylation of the resulting carbon ion.

CONFIDENTIAL

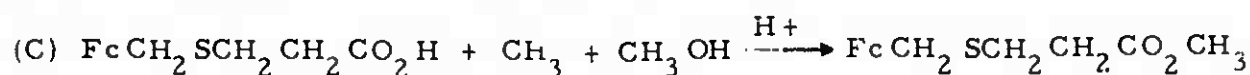


Starting material was recovered.

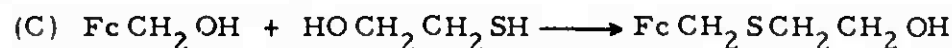
(C) β -(Ferrocenylmethylthio) propionic acid (m. p. , 80 to 81°) was prepared by reaction of hydroxymethyl ferrocene with β -mercaptopropionic acid in water.



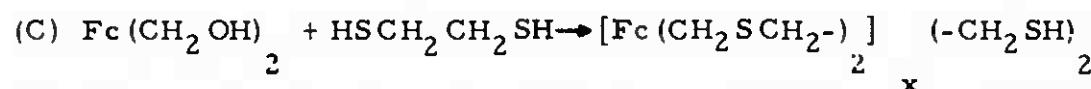
(C) Esterification with methanol afforded methyl β -(ferrocenylmethylthio) propionate [b. p. , 260° C (DTA)] in good yield.



(C) Reaction of hydroxymethyl ferrocene with mercaptoethanol in water gave β -(ferrocenylmethylthio) ethanol (b. p. , 290° C) in high yield.



(C) The condensation polymerization of 1,1'-dihydroxymethyl ferrocene with a slight excess of 1,2-dimercaptoethane in water yielded mercaptomethyl terminated poly- [bis (methylthiomethyl) ferrocene] (denoted poly F S -1) (m. p. , 119 to 122° C).



(U) The structural assignments of the above catalysts were confirmed by infrared analysis. The physical constants are as follows:

CONFIDENTIAL

CONFIDENTIAL

Compound	M. p., ° C	B. p., ° C	Theor. % Fe
Ethoxymethyl Ferrocene	---	88/0.15	23
Bisferrocenylmethyl Ether	126-129	---	27
Allyloxymethyl Ferrocene	---	260	22
Bis (α ferrocenylethyl) Ether	---	197, 232 (dec)	25
Ferrocenylmethyl acetate	75-77	---	22
Ferrocenylmethyl ferrocenoate	131-135	---	26
β - (Ferrocenylmethylthio) propionic acid	80-81	---	18
Methyl β - (ferrocenylmethylthio) propionate	260 (DTA)	---	18
β - (Ferrocenylmethylthio) ethanol	290 (DTA)	---	20
poly F S -1	119-122	---	16
Methyl ferrocenylacetate	---	282	22
Ethyl ferrocenylacetate	---	286	21
n-Butyl ferrocenylmethyl ferrocene	---	274, 300 (dec)	19

CONFIDENTIAL

(U) It has been determined that the freezing points of compounds recorded in the first quarterly technical report (AFRPL-TR-67-99, March 1967) are in error. Many of the compounds froze when held for a number of hours at temperatures substantially higher than their reported freezing points. The freezing points were obtained by conventional methods. Failure to reach equilibrium at moderate rates of cooling (ca. 2°/min.) can possibly be attributed to the high viscosity of these compounds at low temperatures and substantial super cooling. Only melting points will be reported in the future, since the melting point of a compound is less time-dependent than the freezing point.

(C) The time dependence of freezing in this type of compound is substantial. For example, PLASTISCAT-IV^(R) has a melting point of 8 to 11° C, but the freezing point was determined to be -25 to -34° C. PLASTISCAT-IV^(R) froze when held at 10° C for a number of hours.

(U) An accurate determination of the melting point of each compound synthesized is now in progress.

(U) Compounds prepared during this program for which analyses have been obtained are as follows:

CONFIDENTIAL

(U) $\text{FcCO}_2\text{CH}_2\text{C} \equiv \text{CH}$ (propargyl ferrocenyl acetate)

(C) Analysis calculated for $\text{C}_{14}\text{H}_{12}\text{FeO}_2$: C, 62.72; H, 4.51; Fe, 20.83.

Found: C, 62.96; H, 4.64; Fe, 21.07.

(U) $\text{FcCO}_2\text{CH}_2\text{CH}_2\text{CH}_3$ (n-propyl ferrocenyl acetate)

(C) Analysis calculated for $\text{C}_{14}\text{H}_{16}\text{FeO}_2$: C, 61.79; H, 5.93; Fe, 20.52.

Found: C, 62.04; H, 6.16; Fe, 19.8.

(U) $\text{FcCH}_2\text{OCH}_3$ (methoxymethyl ferrocene)

(C) Analysis calculated for $\text{C}_{12}\text{H}_{14}\text{FeO}$: C, 62.64; H, 6.13; Fe, 24.27.

Found: C, 62.94; H, 6.30; Fe, 24.55.

(U) $\text{CH}_3\text{OCH}_2\text{FcCH}_2\text{OH}$ (1-hydroxymethyl-1-methoxymethyl ferrocene)

(C) Analysis calculated for $\text{C}_{13}\text{H}_{17}\text{FeO}_2$: C, 60.05; H, 6.20; Fe, 21.48.

Found: C, 59.72; H, 6.46; Fe, 21.10.

(U) $\text{Fc}(\text{CH}_2\text{OCH}_3)_2$ [1,1'-di(methoxymethyl) ferrocene]

(C) Analysis calculated for $\text{C}_{14}\text{H}_{19}\text{FeO}_2$: C, 61.36; H, 6.62; Fe, 20.38.

Found: C, 61.31; H, 6.81; Fe, 20.21.

(U) $\text{CH}_3\text{CO}_2\text{CH}_2\text{Fc}$ (ferrocenyl methyl acetate)

(C) Analysis calculated for $\text{C}_{13}\text{H}_{14}\text{FeO}_2$: C, 60.49; H, 5.47; Fe, 21.64.

Found: C, 60.75; H, 5.56; Fe, 21.95.

(U) $\text{FcCH}_2\text{CO}_2\text{CH}_2\text{CH}_3$ (ethyl ferrocenylacetate)

(C) Calculation for $\text{C}_{14}\text{H}_{16}\text{FeO}_2$: C, 62.25; H, 5.22; Fe, 20.68.

Found: C, 62.59; H, 5.58; Fe, 20.94.

(U) $\text{FcCH}_2\text{S}(\text{CH}_2)_3\text{CH}_3$ (n-butylthiomethyl ferrocene)

(C) Analysis calculated for $\text{C}_{15}\text{H}_{20}\text{FeS}$: C, 62.50; H, 6.99; Fe, 19.38.

Found: C, 62.18; H, 6.86; Fe, 19.50.

CONFIDENTIAL

(U) $C_{14}H_{16}FeO$ (allyloxy methyl ferrocene)

(C) Calculated for $C_{14}H_{16}FeO$: C, 65.65; H, 6.30.

Found: C, 65.80; H, 6.58.

(U) $FcCH_2OCH_2CH_3$ (ethoxymethyl ferrocene)

(C) Analysis calculated for $C_{13}H_{16}FeO$:

C, 63.96; H, 6.67; Fe, 22.88; Mol. Wt., 244.

Found: C, 64.06; H, 6.72; Fe, 22.84; Mol. Wt., 240.

(U) $FcCH(CH_3)OCH(CH_3)Fc$ [Bis(η -ferrocenylethyl) ether]

(C) Analysis calculated for $C_{24}H_{26}Fe_2O$: C, 65.19; H, 5.93; Fe, 25.26.

Found: C, 65.47; H, 6.09; Fe, 25.23.

(U) During this reporting period effort was expended to prepare substantial amounts of hydroxymethyl ferrocene. This compound is a necessary precursor to many of the more promising candidates, and at present the pure compound cannot be purchased at a reasonable price. To date 0.5-pound of this compound has been prepared.

(U) 2. Phase II - Decomposition Studies and Catalyst Evaluation

(U) Upon completion of synthesis, elemental characterization and physical property determination of each candidate catalyst, an evaluation of the compatibility of the candidate catalysts with other propellant ingredients will be accomplished. Following the compatibility, the actual effectiveness of the candidate catalysts will be measured by determining the propellant processing characteristics as well as the burning rate.

(U) a. Catalyst Compatibility Studies

(C) Compatibility testing at ambient temperatures and at 160° F between catalyst candidates and ammonium perchlorate, CTPB binder, curing agents and aluminum were reported in Quarterly Technical Report AFRPL-TR-67-99 (March 1967). In summary, no serious incompatibility problems were noted with the exception of dimethylaminomethyl ferrocene, which at mix temperature (120 - 160° F), reacted with the oxidizer to liberate ammonia and served as a base catalyst for the exothermic polymerization of MAPO.

CONFIDENTIAL

(U) The major effort during this report period was to test the compatibility of ammonium perchlorate and candidate catalysts at elevated temperatures (mixing and cure temperatures) for periods up to ten days. The differences noted in the differential thermal analyses (DTA) of oxidizer-catalyst mixtures after 10 days gave a strong indication as to whether catalyst oxidation could possibly occur during a cure cycle. It should be pointed out that the test is considerably more severe than that which would be encountered in an actual propellant. In essence, it was believed that if ignition or at least exothermic decomposition did not occur in these tests, no problem should be expected in the propellant.

(C) Ammonium perchlorate containing 5 percent catalyst (by weight) was subjected to a temperature of 160° F for ten days. The catalysts evaluated are those listed in Table I. Visual observations show no apparent change in color of any catalyst/ammonium perchlorate mixture, except for methoxymethyl ferrocene, which apparently oxidized extensively, and 2 - methoxyethyl ferrocenate, which changed color from orange to lemon. Microscopic examinations showed particles of red iron oxide (Fe_2O_3) in the methoxymethyl ferrocene/ammonium perchlorate mixture and the color change of 2 - methoxyethyl ferrocenate may simply be attributed to dilution by ammonium perchlorate.

(U) Ammonium perchlorate standard grind (50-55 μ weight mean diameter) was selected to provide more intimate surface contact, especially with the solid catalysts, and thus provide a more severe test of compatibility.

(C) After ten days at a temperature of 160° F, a differential thermal analysis of these oxidizer catalyst mixtures was obtained. A control sample of standard grind ammonium perchlorate was subjected to the same conditions as the catalyst/oxidizer mixtures (Figure 1). The DTA plot of trimethoxy iron exhibited no change from original unaged values. The remainder of the catalysts examined seemingly showed some signs of interaction. Probably the most significant change was noted with methoxymethyl ferrocene (Figure 2). The reduction in intensity of the peak at 182° C supports the previously mentioned postulation that oxidation has occurred. In essence, the peak at 182° C can be attributed to oxidation of the methoxymethyl ferrocene by the small amount of oxidizer decomposition gases present at 160° F. Figures 3 through 13 show the differential thermal analyses of the rest of the oxidizer/catalyst mixtures. At this point, it is believed that many of the variations observed following aging can be attributed mainly to slight oxidation and improved dispersion of the catalyst in the oxidizer during the aging process. No chemical explanation is available at this time for some of the new peaks and disappearance of others.

(C) In order to determine how serious these interactions might be and to ascertain the degree of the interaction, the catalyst, after being in contact with the oxidizer for 10 days at 160° F, was subjected to thin layer chroma-

CONFIDENTIAL

CONFIDENTIAL

CONFIDENTIAL

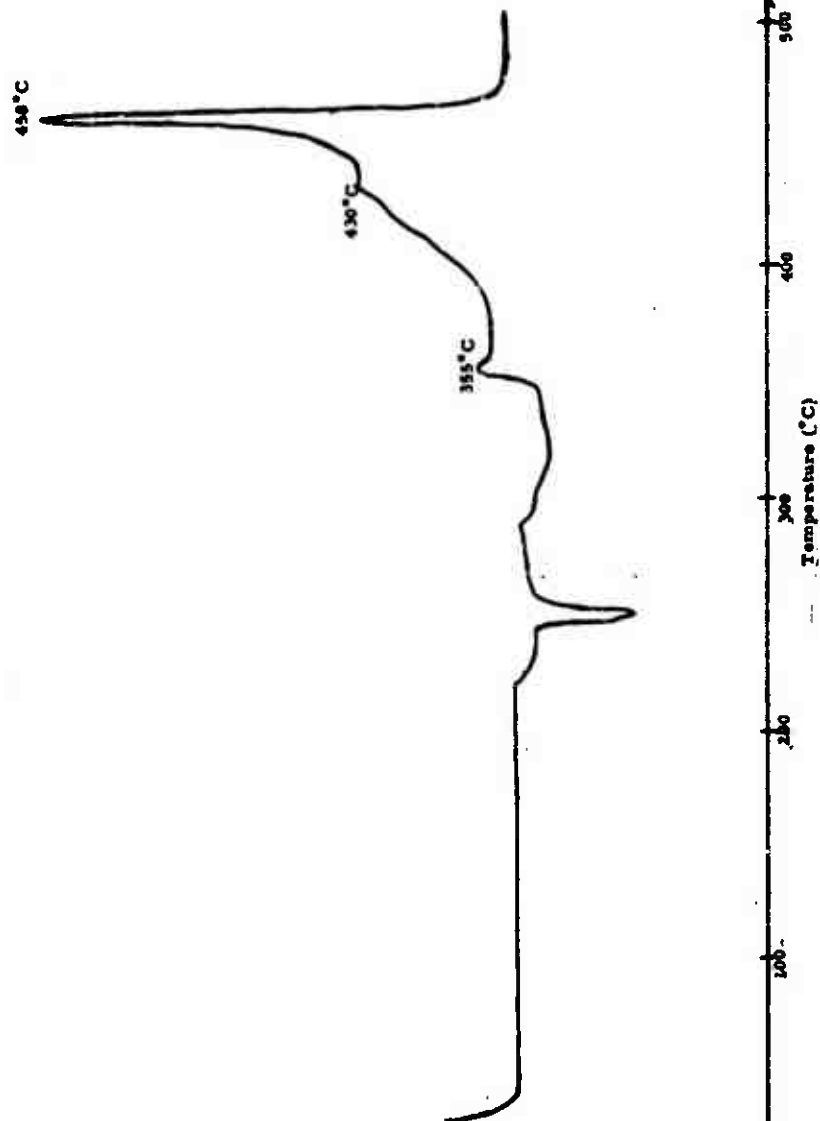


Figure 1. DTA of Standard Grind AP (50-55 Micron WMD) Control After 10 Days at 160°F.

CONFIDENTIAL

CONFIDENTIAL

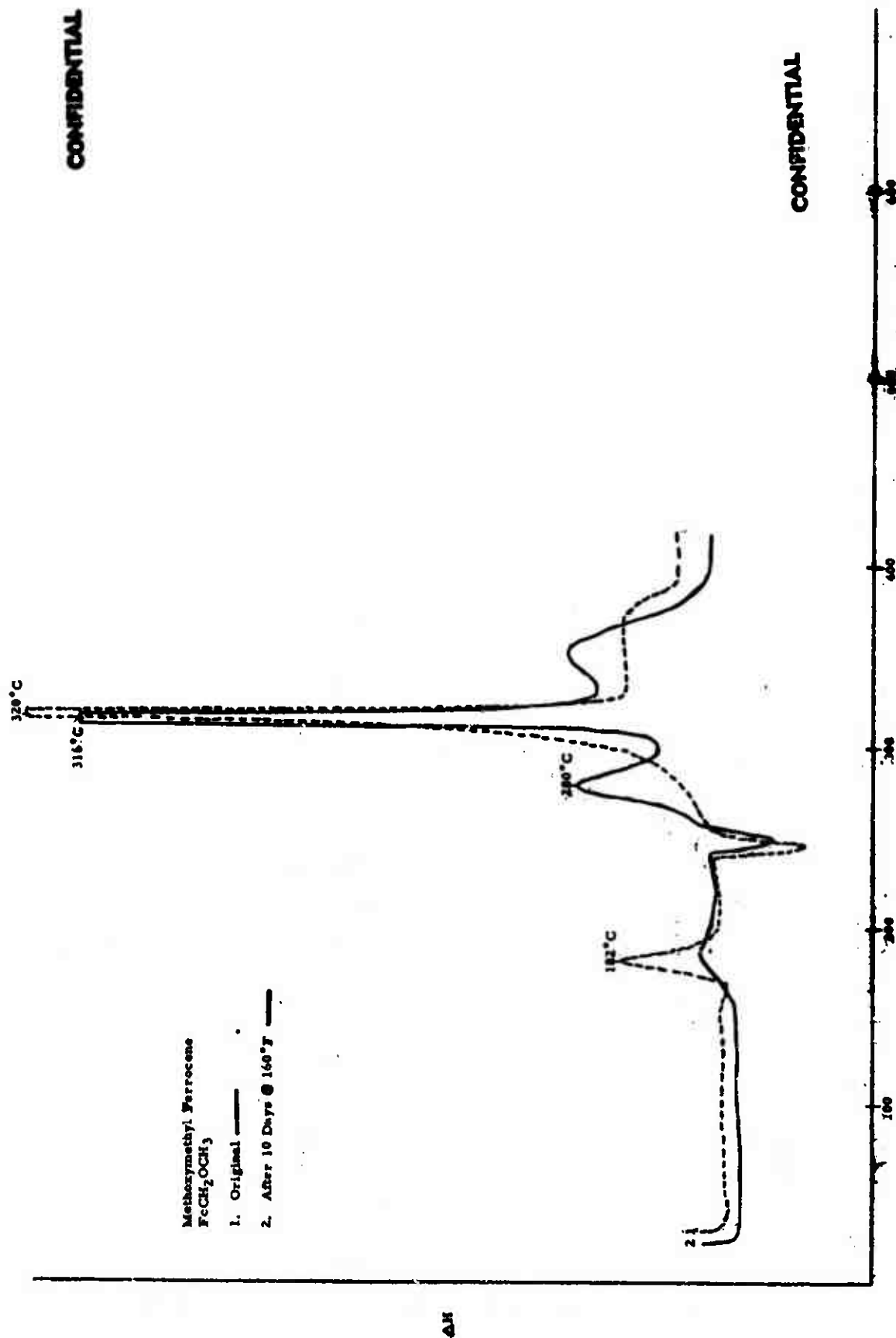


Figure 2. DTA of AP (Standard Grind) + 5 Percent Catalyst (Methoxymethyl Ferrocene).

CONFIDENTIAL

CONFIDENTIAL

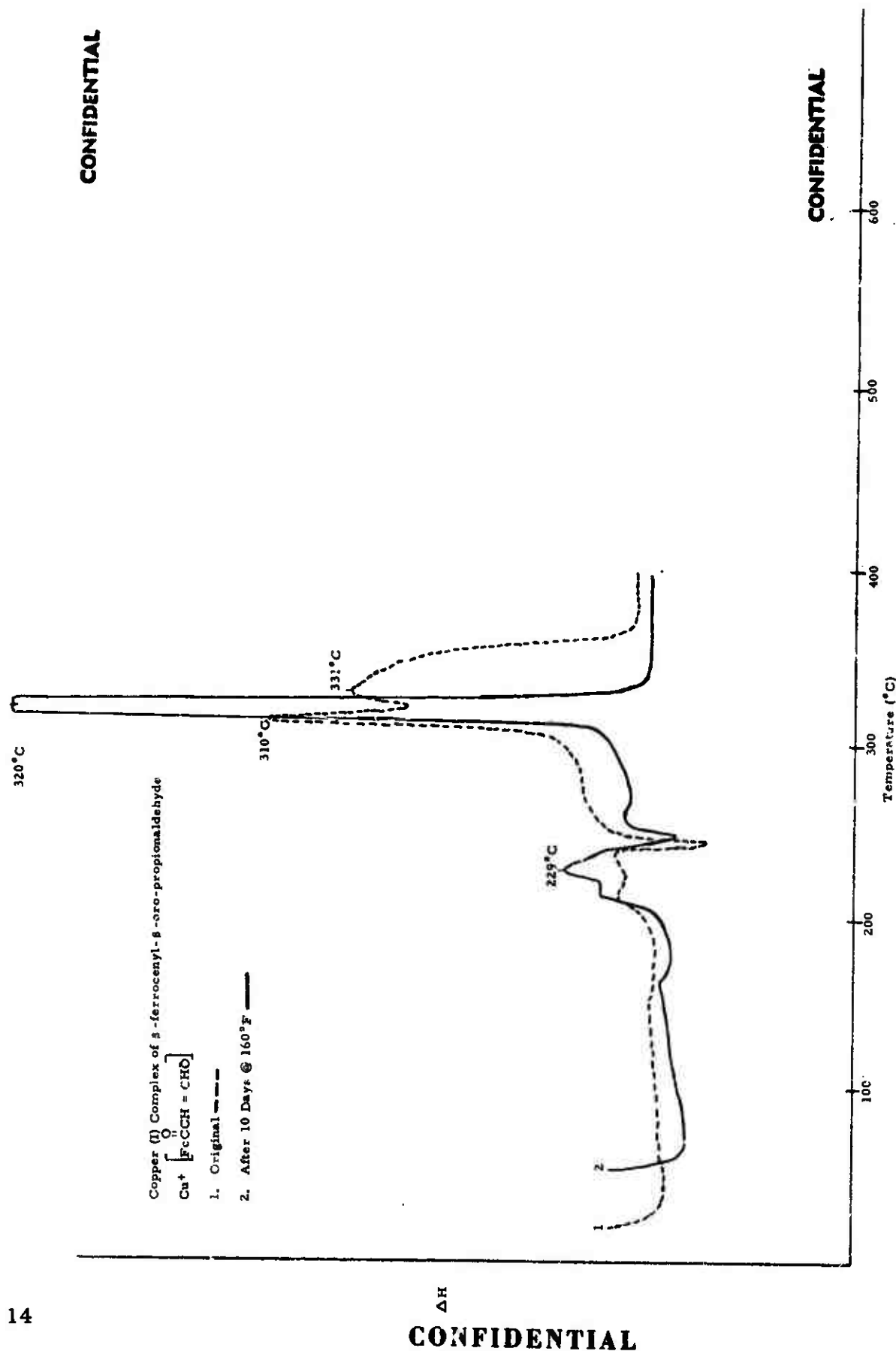


Figure 3. DTA of AP (Standard Grind) + 5 Percent Catalyst - Copper (I) Complex of β -Ferrocenyl- β -oxo-propionaldehyde.

CONFIDENTIAL

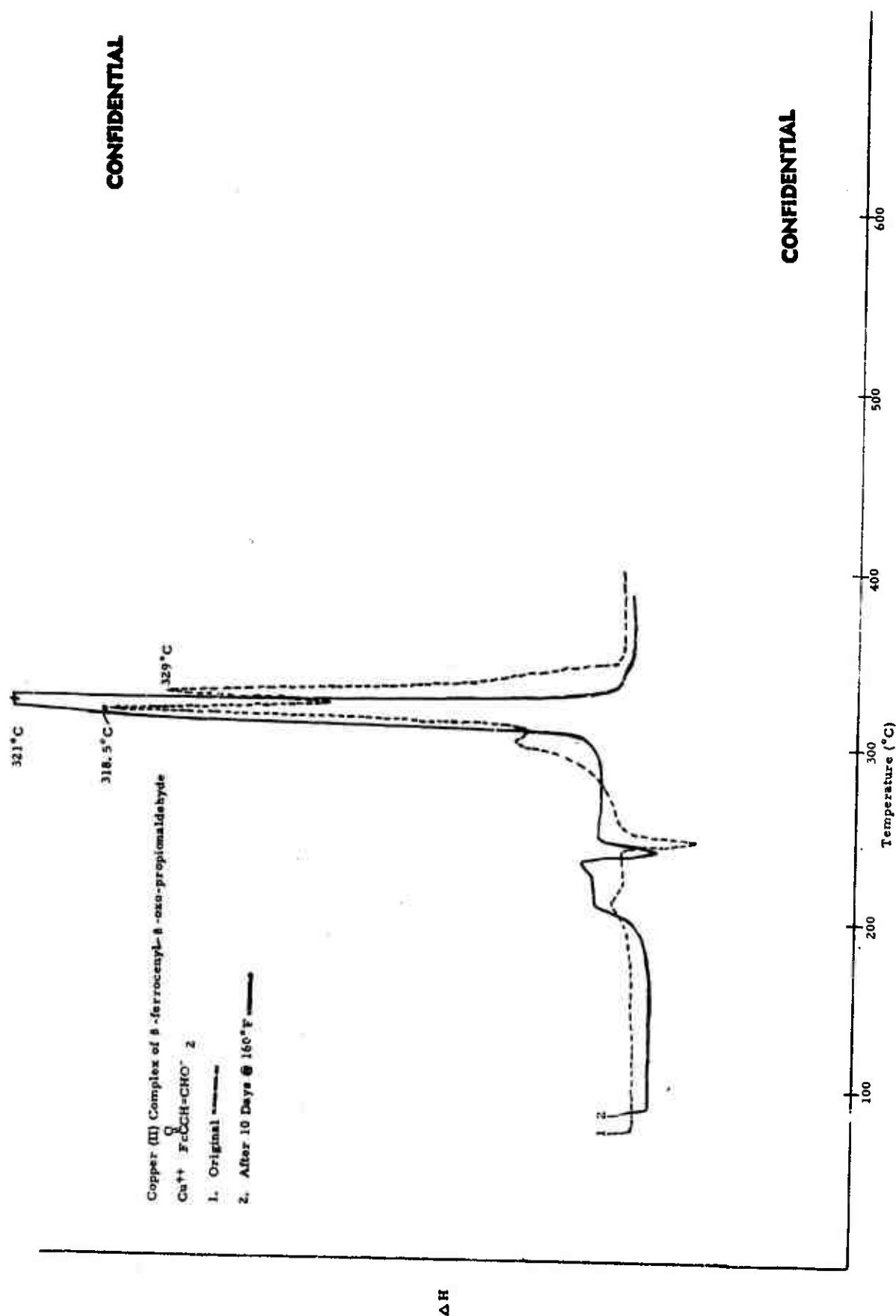


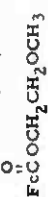
Figure 4. DTA of Copper (II) Complex of β-Ferrocenyl-β-oxo-propionaldehyde.

CONFIDENTIAL

CONFIDENTIAL

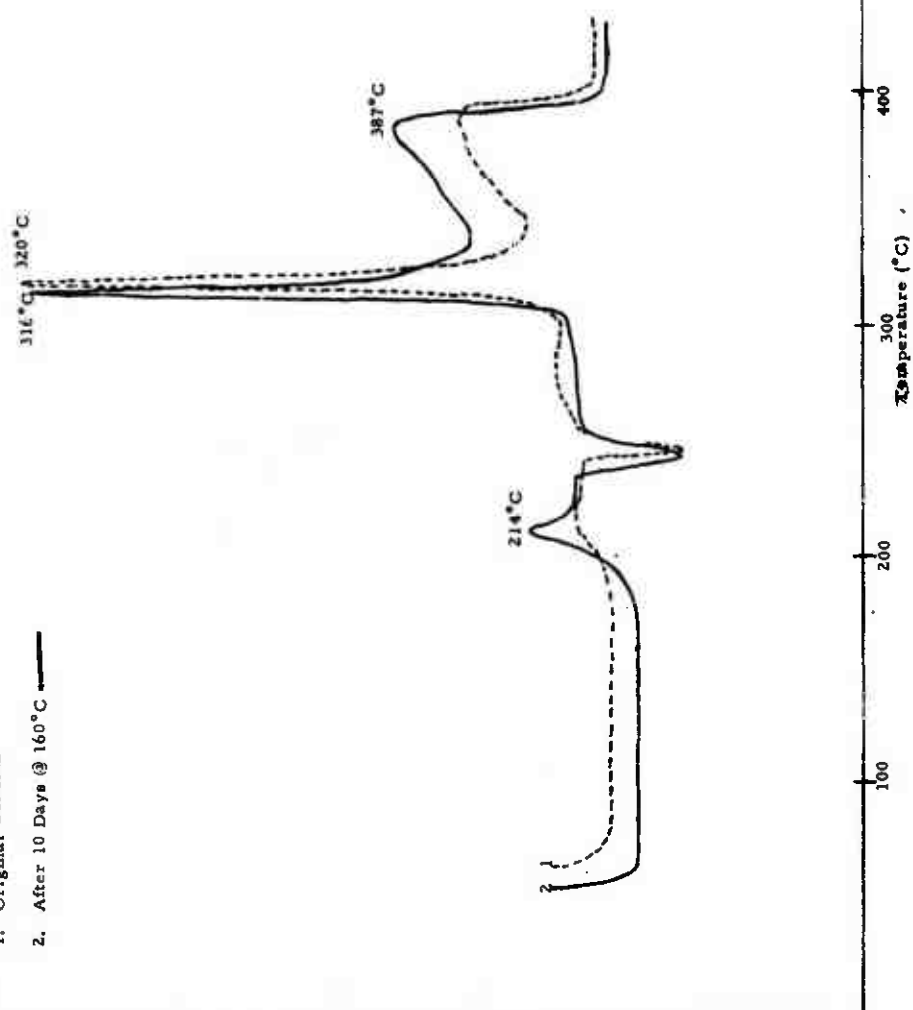
CONFIDENTIAL

2-Methoxyethyl Ferrocenecarboxylate



1. Original - - - - -

2. After 10 Days @ 160°C - - - - -

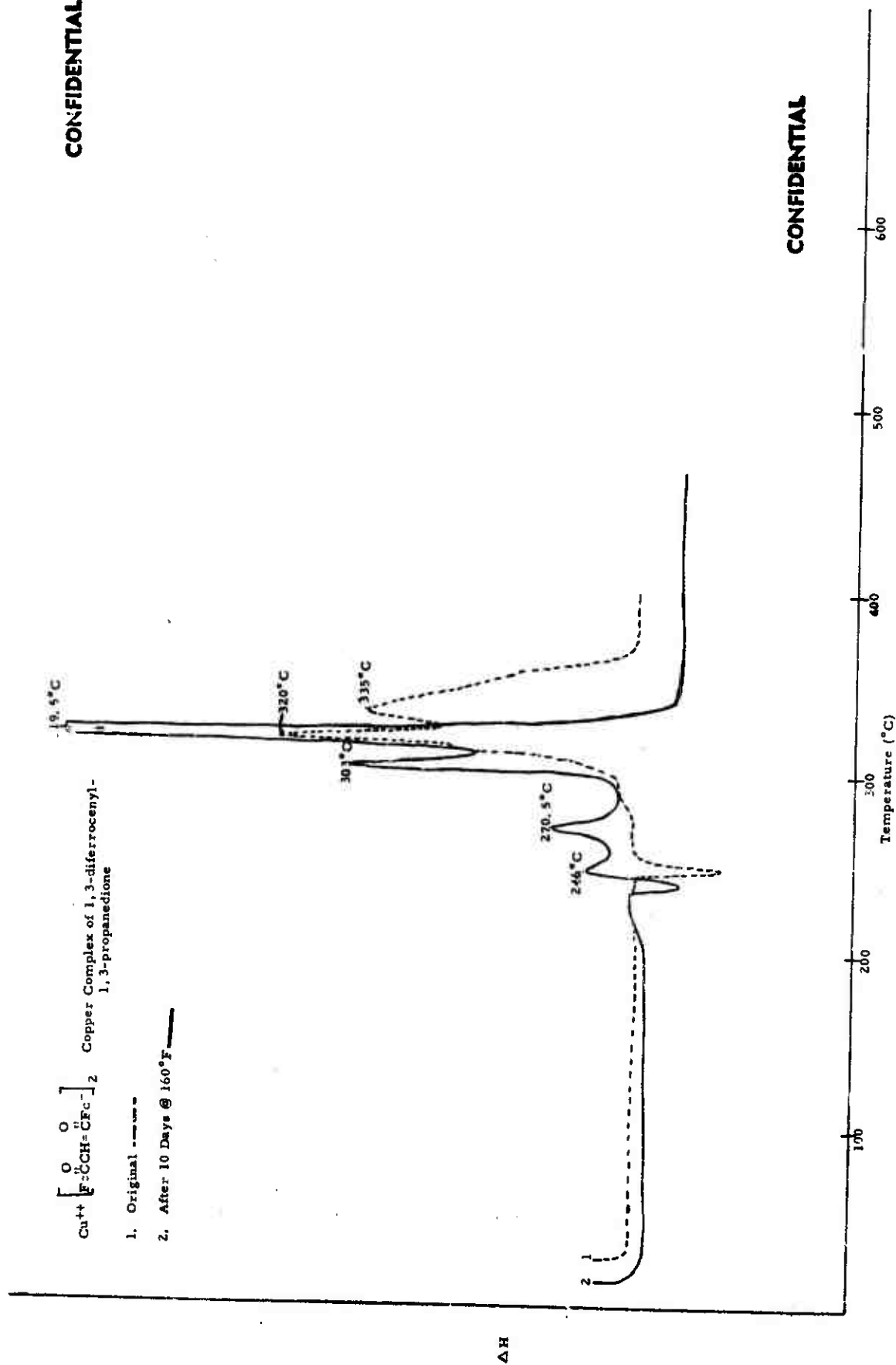


CONFIDENTIAL

Figure 5. DTA of 2-Methoxyethyl Ferrocenecarboxylate.

CONFIDENTIAL

CONFIDENTIAL



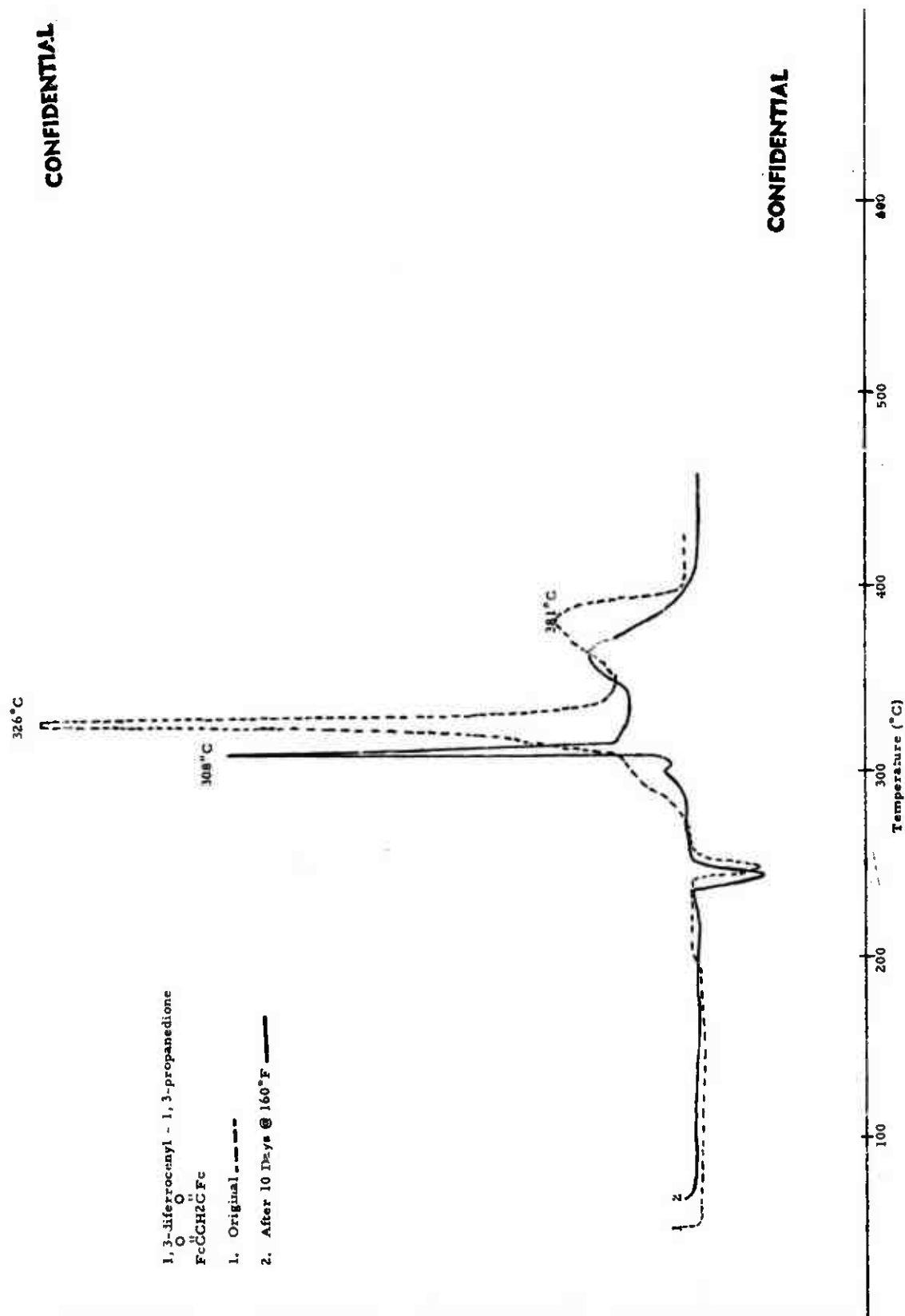
CONFIDENTIAL

Figure 6. DTA of Copper Complex of 1,3-diferrocenyl-1,3-propanedione.

CONFIDENTIAL

CONFIDENTIAL

CONFIDENTIAL



CONFIDENTIAL

Figure 7. DTA of 1,3-diferrocenyl-1,3-propanedione.

CONFIDENTIAL

CONFIDENTIAL

CONFIDENTIAL

CONFIDENTIAL

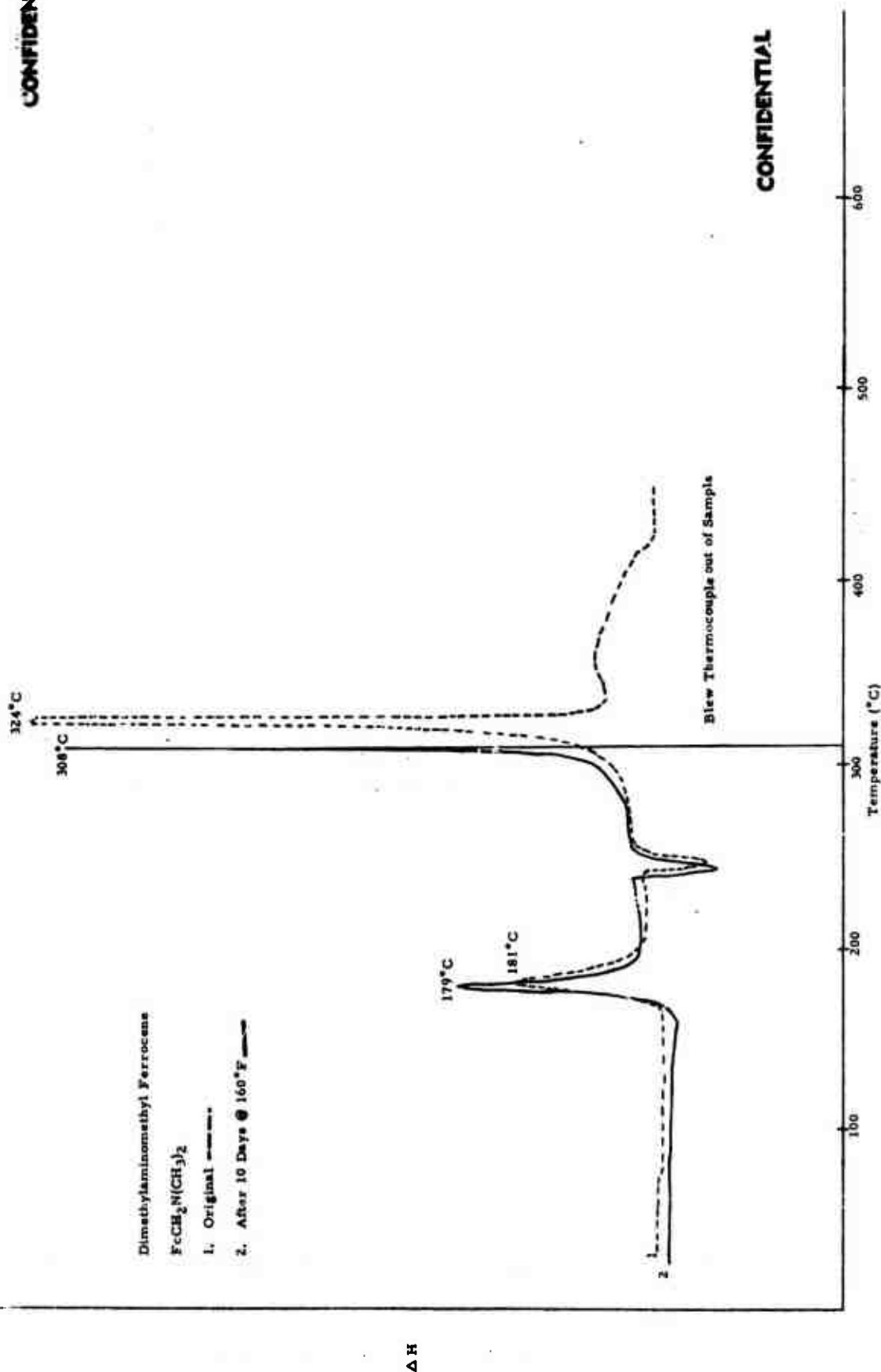


Figure 8. DTA of dimethylaminomethyl ferrocene.

CONFIDENTIAL

CONFIDENTIAL

CONFIDENTIAL

CONFIDENTIAL

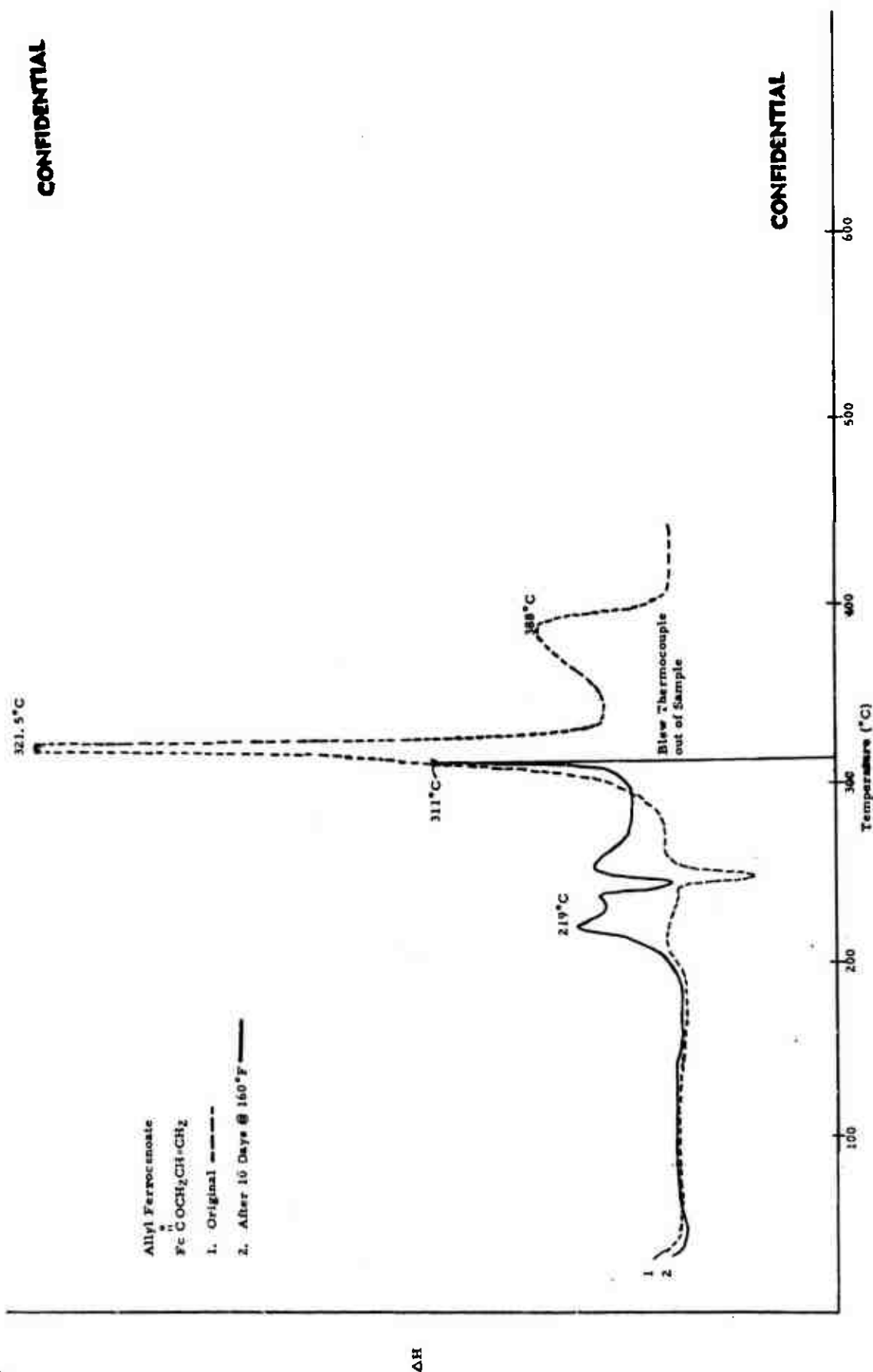


Figure 9. DTA of allyl ferrocenoate.

CONFIDENTIAL

CONFIDENTIAL

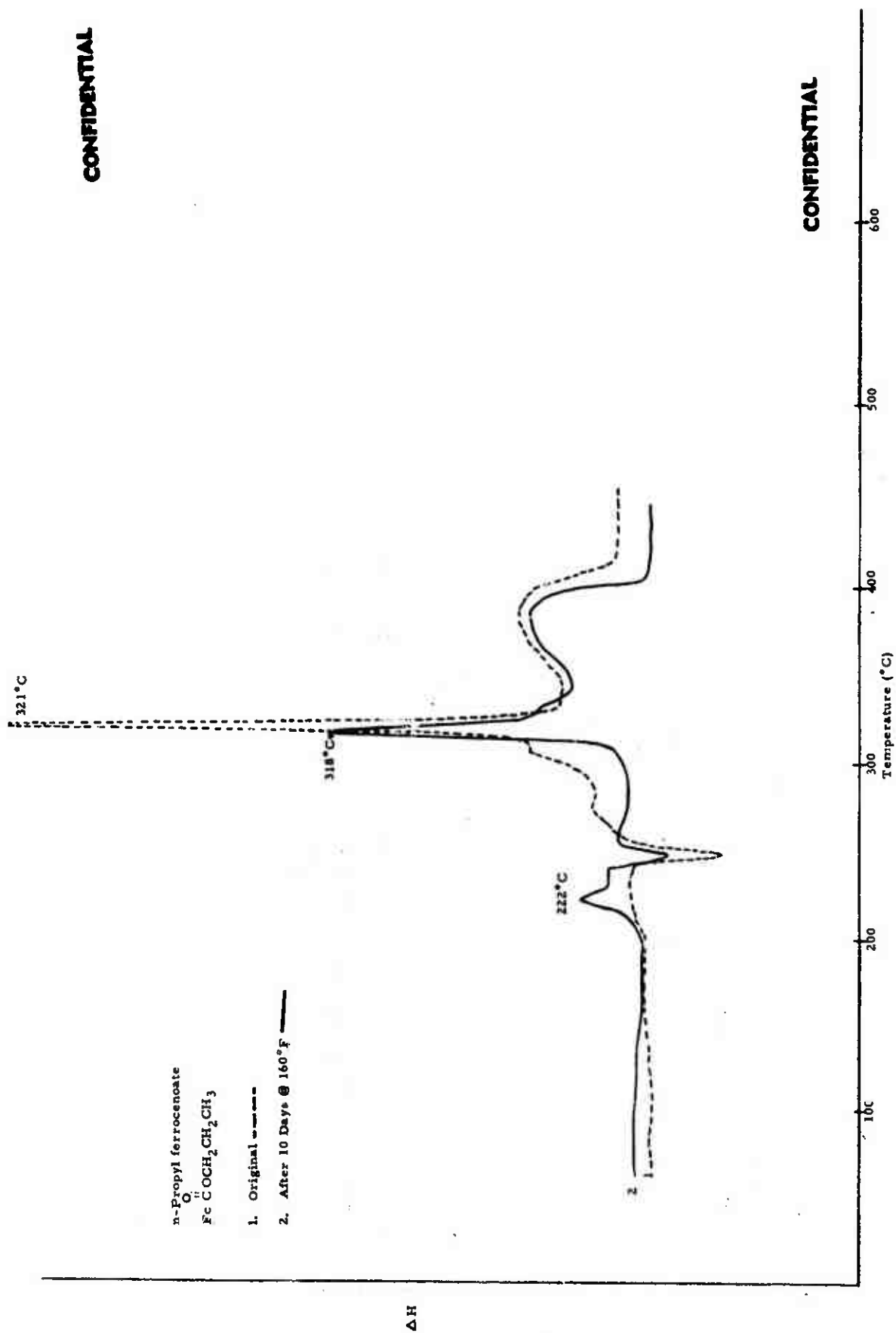


Figure 10. DTA of n-propyl ferrocenolate.

CONFIDENTIAL

CONFIDENTIAL

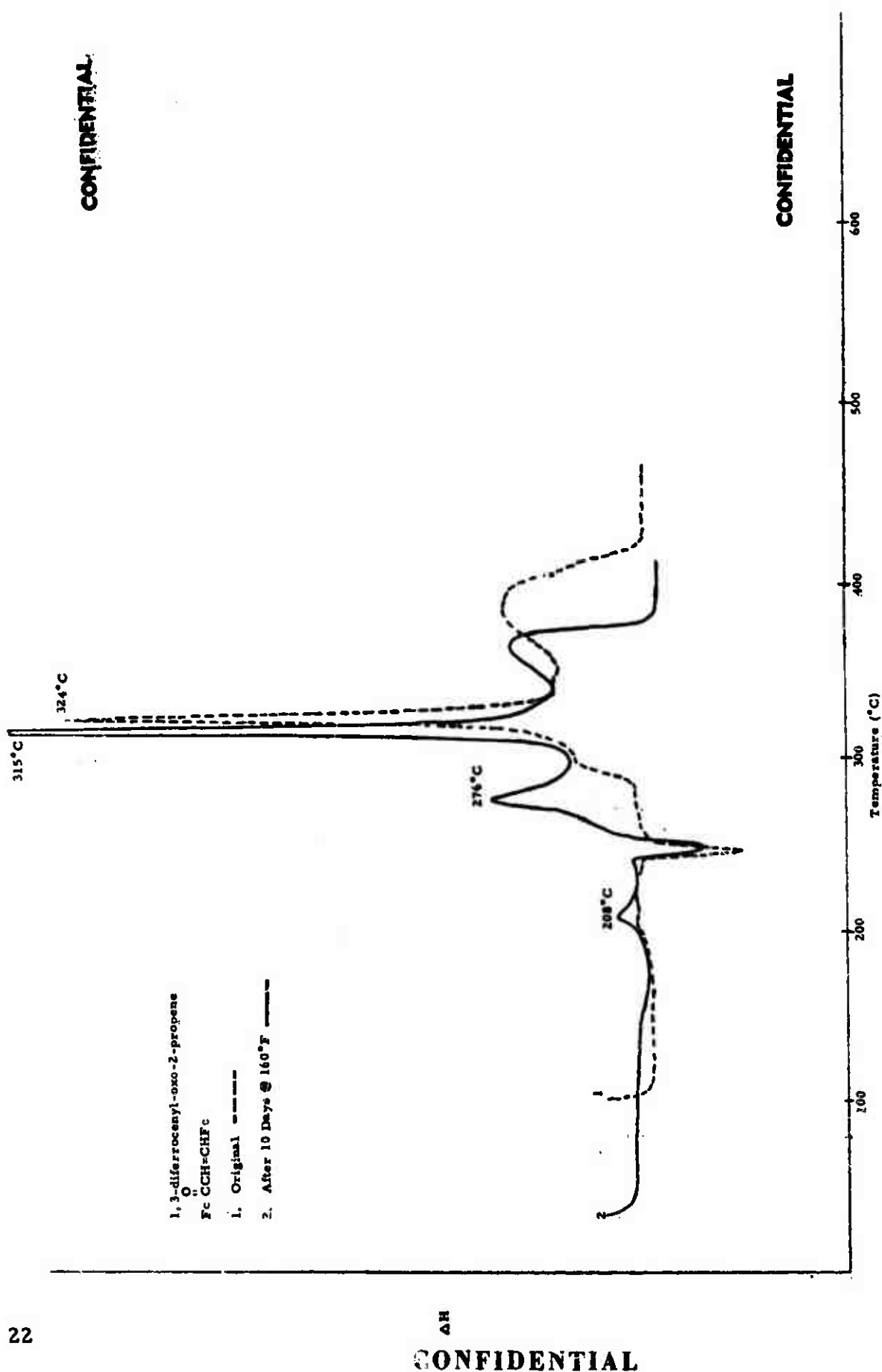
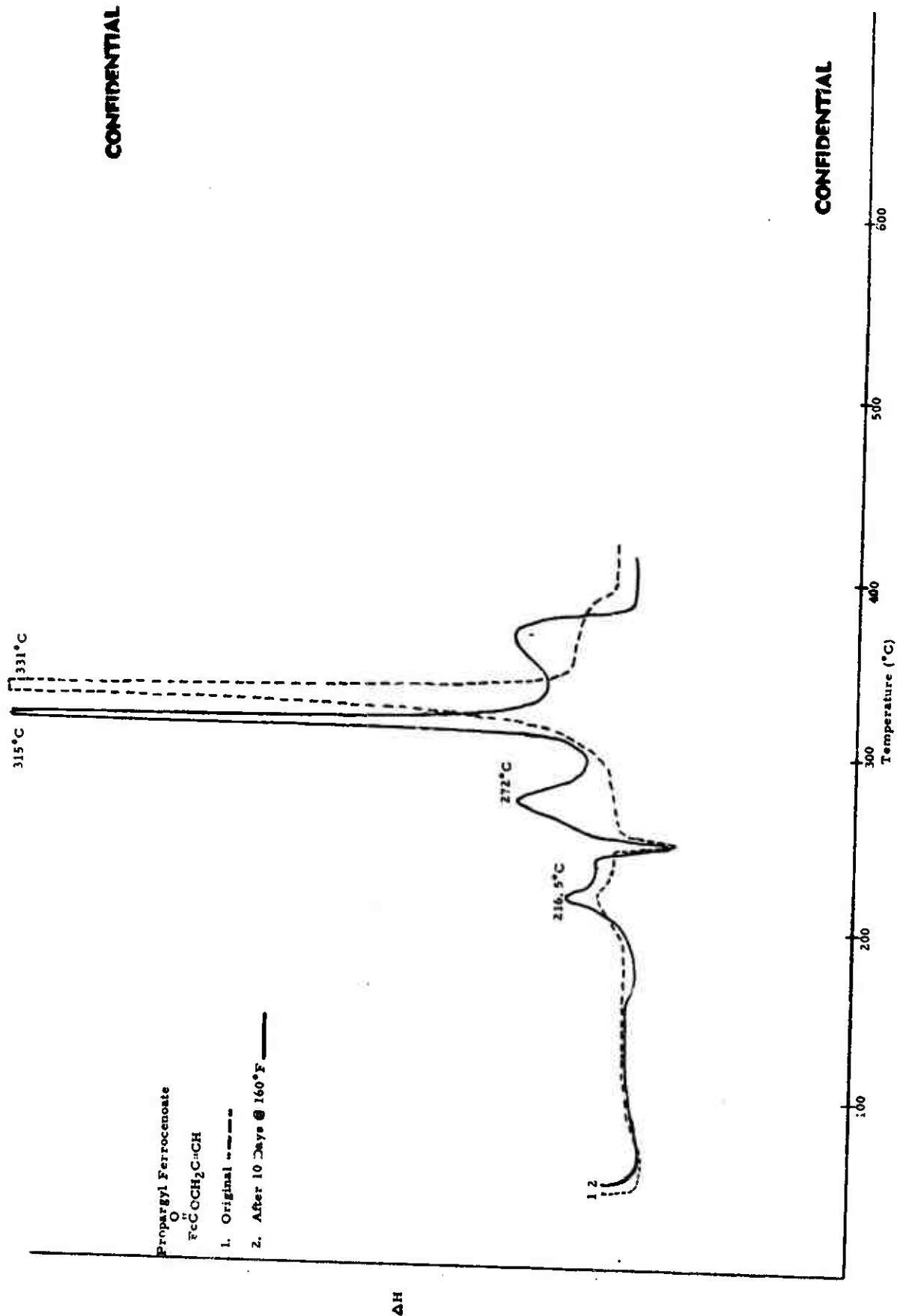


Figure 11. DTA of 1,3-diferrocenyl-oxo-2-propene.

CONFIDENTIAL

CONFIDENTIAL



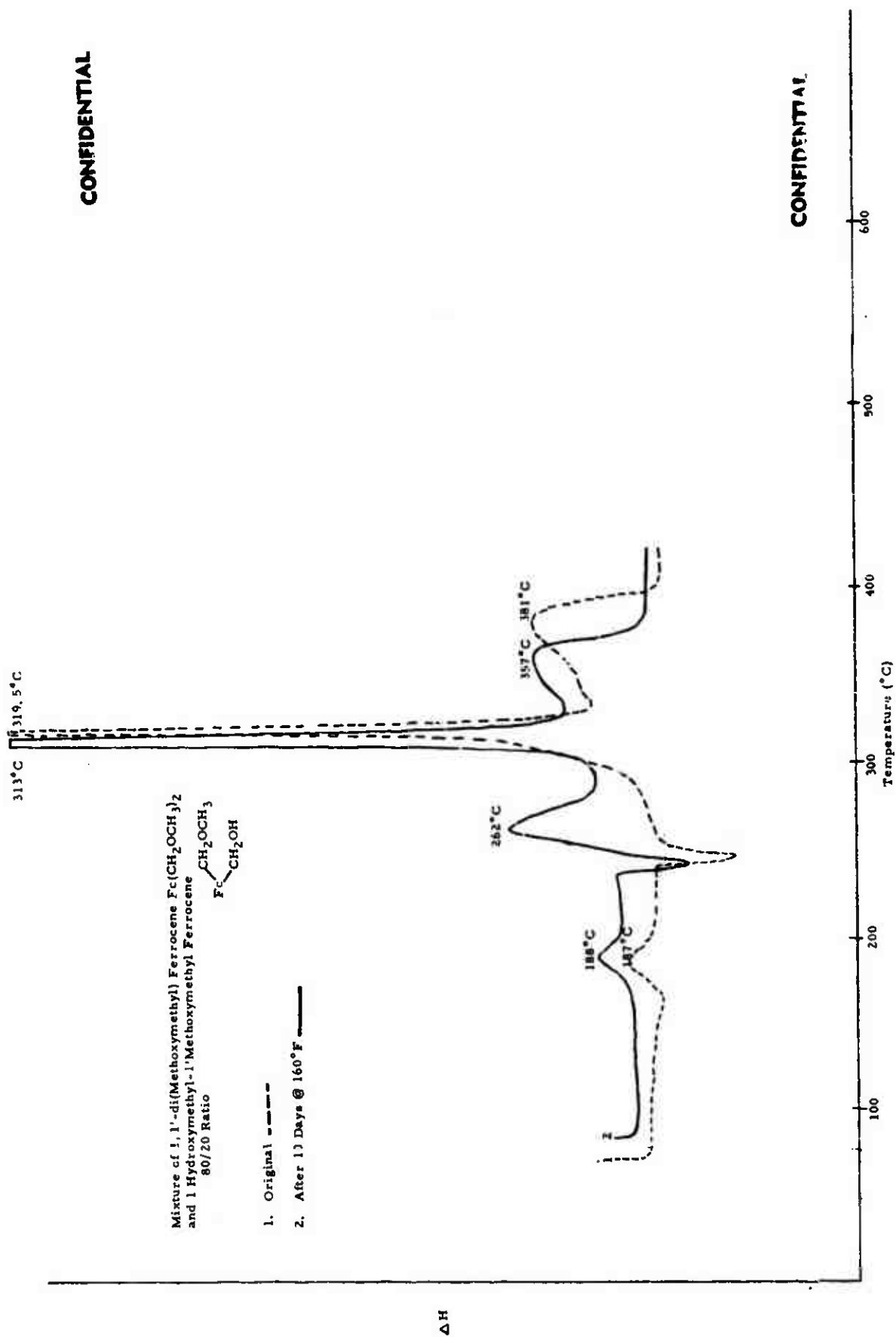
CONFIDENTIAL

Figure 12. DTA of propargyl ferrocenoate.

CONFIDENTIAL

CONFIDENTIAL

CONFIDENTIAL



CONFIDENTIAL

Figure 13. DTA of Mixture of 1,1'-di(methoxymethyl) ferrocene and 1 hydroxymethyl-1'-methoxymethyl ferrocene.

CONFIDENTIAL

CONFIDENTIAL

tography. The catalysts evaluated and the results of this evaluation by thin-layer chromatography (TLC) are shown in Table II. Thin-layer chromatography substantiated conclusions previously drawn that methoxymethyl ferrocene and the mixture of 1,1'-di (methoxymethyl) ferrocene with 1-hydroxymethyl-1'-methoxymethyl ferrocene also oxidized considerably. In general, TLC shows that some catalysts changed slightly, possibly because of heat or the presence of oxidizer, or both, and some trace impurities may have decomposed.

(C) Sensitivity to impact and friction was determined for a mixture of ammonium perchlorate and catalyst (5 percent by weight) which had been aged at 160° F for 10 days. The results are summarized in Table III and are compared to non-catalyzed ammonium perchlorate and n-butyl ferrocene catalyzed oxidizer. All the catalysts compared quite favorably with n-butyl ferrocene² (P-IV(R)) catalyzed oxidizer (5 percent).

(C) Some newly synthesized catalysts [methyl- β -(ferrocenylmethylthio) propionate, ferrocenylmethyl ferrocenoate, β -(ferrocenylmethylthio) ethanol] and n-butyl thiomethyl ferrocene were evaluated by DTA during this report period. These compounds do not appear to affect the decomposition of ammonium perchlorate differently than those previously tested. A plot of these data is shown on Figures 14 and 15.

(C) Figure 16 is a composite DTA plot of other recently synthesized catalysts and shows no radical difference in the decomposition of ammonium perchlorate over those catalysts evaluated previously.

(C) A tabulation of the temperatures at the major peak decomposition are as follows:

CONFIDENTIAL

	<u>Temperature</u> (°C)
a. Allyloxymethyl ferrocene	320
b. Ethoxymethylferrocene	318.5
c. Bis-ferrocenylmethyl ether	320
d. Methyl ferrocenylacetate	316
e. Bis-(α -ferrocenylethyl) ether	324
f. Ethyl ferrocenylacetate	315.5
g. Methyl- β (ferrocenylmethylthio) propionate	326
h. Ferrocenylmethyl ferrocenoate	320
i. β -(ferrocenylmethylthio) ethanol	317
j. n-butyl thiomethyl ferrocene	322

CONFIDENTIAL

2. U. S. Patent Secrecy Order applies.

TABLE II

CONFIDENTIAL

CATALYST COMPATIBILITY THIN-LAYER CHROMATOGRAPH RESULTS (160°F)

Catalyst	Original Values	Values after 10 days
$\text{Fe}(\text{OCH}_3)_3$	Spot at R_f 0, Trace at R_f 0.7	Spot at R_f 0, Trace at R_f 0.7
$\text{FeCH}_2\text{OCH}_3$	Spot at R_f 0, Trace at R_f 0.7, 0.8, 0.9	{Spot at R_f 0, R_f 0.1 - 0.2 (90%) R_f 0.2 - 0.5 (9-10%) Trace at R_f 0.5 - 0.8
$\text{Cu}^+ [\text{Fc} \overset{\text{O}}{\underset{\text{O}}{\text{C}}} \text{CH} = \text{CH} \overset{\text{O}}{\underset{\text{O}}{\text{C}}}]$	Spot at R_f 0, Streaked R_f 0 - 0.5, Spot R_f 0.7	Spot R_f 0, Badly streaked from R_f 0 - 0.4
$\text{Cu}^{++} [\text{Fc} \overset{\text{O}}{\underset{\text{O}}{\text{C}}} \text{CH} = \text{CH} \overset{\text{O}}{\underset{\text{O}}{\text{C}}}]_2$	Streaked R_f 0 - 0.5, Trace at R_f 0.7	{Spot at R_f 0.4 (80%) Spot at R_f 0.5 (90%) Spot at R_f 0.6
$\text{Fc} \overset{\text{O}}{\underset{\text{O}}{\text{C}}} \text{COCH}_2\text{CH}_2\text{OCH}_3$	Spot at R_f 0.1, 0.5, 0.7	Spot at R_f 0, badly streaked from R_f 0 - 0.3
$\text{Cu}^{++} [\text{Fc} \overset{\text{O}}{\underset{\text{O}}{\text{C}}} \text{CH} \overset{\text{O}}{\underset{\text{O}}{\text{C}}} \text{Fc}]_2$	Spot at R_f 0, R_f 0.1	Spot at R_f 0.3 (50%)
$\text{Fc} \overset{\text{O}}{\underset{\text{O}}{\text{C}}} \text{CH}_2 \overset{\text{O}}{\underset{\text{O}}{\text{C}}} \text{Fc}$	Spot at R_f 0, R_f 0.2	Spot at R_f 0, dual spot at R_f 0.4 (80/10 ratio)
$\text{FcCH}_2\text{N}(\text{CH}_3)_2$	Spot at R_f 0, R_f 0.3	{Spot at R_f 0.05 (20%) and R_f 0.1 (80%) Trace at R_f 0.7
$\text{Fc} \overset{\text{O}}{\underset{\text{O}}{\text{C}}} \text{COCH}_2\text{CH} = \text{CH}_2$	Trace at R_f 0	Trace at R_f 0, spot at R_f 0.7
$\text{Fc} \overset{\text{O}}{\underset{\text{O}}{\text{C}}} \text{COCH}_2\text{CH}_2\text{CH}_3$	Spot at R_f 0, dual spot at R_f 0.6	Trace at R_f 0, spot at R_f 0.7
$\text{Fc} \overset{\text{O}}{\underset{\text{O}}{\text{C}}} \text{CH} = \text{CHFc}$	Trace at R_f 0, spot at R_f 0.5	Trace at R_f 0, dual spot at R_f 0.6 90/10 ratio
$\text{Fc} \overset{\text{O}}{\underset{\text{O}}{\text{C}}} \text{COCH}_2\text{C} = \text{CH}$	Trace at R_f 0, spot at R_f 0.5	Trace at R_f 0, spot at R_f 0.6
Mixture of $\text{Fc}(\text{CH}_2\text{OCH}_3)_2$ and $\text{Fc} \begin{array}{c} \text{CH}_2\text{OCH}_3 \\ \diagup \quad \diagdown \\ \text{Fc} \quad \text{CH}_2\text{OH} \end{array}$	Spot at R_f 0, trace at R_f 0.3, spot at R_f 0.4	Spot at R_f 0, trace at R_f 0.6

CONFIDENTIAL

CONFIDENTIAL

CONFIDENTIAL

TABLE III

IMPACT AND FRICTION SENSITIVITY OF CATALYST/AMMONIUM PERCHLORATE(5%)

Catalyst	$\frac{F_o}{(rpm)}$	$\frac{E_o}{(Kg-cm)}$
Trimethoxy Iron	>6000	100 +
Methoxymethyl ferrocene	>6000	50
Copper (I) complex of β -ferrocenyl- β -oxo-propionaldehyde	>6000	70
Copper (II) complex of β -ferrocenyl- β -oxo-propionaldehyde	>6000	100 +
2-Methoxyethyl ferrocenoate	>6000	100 +
Copper complex of 1, 3, -diferrocenyl-1, 3-propanedione	>6000	100 +
1, 3-diferrocenyl-1, 3-propanedione	>6000	100 +
Dimethylaminomethyl ferrocene	>6000	60
Allyl ferrocenoate	>6000	70
n-propyl ferrocenoate	>4000	70
1, 3-diferrocenyl-1 oxo-2 propene	>6000	100 +
Propargyl ferrocenoate	>6000	100 +
Mixture of 1, 1'-di(methoxymethyl) ferrocene and 1 hydroxymethyl-1'-methoxymethyl ferrocene	>6000	70
$Fc(CH_2OCH_3)_2$ & Fc $\begin{matrix} \swarrow CH_2OCH_3 \\ \searrow CH_2OH \end{matrix}$		
Ammonium perchlorate	>6000	100 +
n-butyl ferrocene [PLASTISCAT-IV (R)]	>6000	50

CONFIDENTIAL

CONFIDENTIAL

CONFIDENTIAL

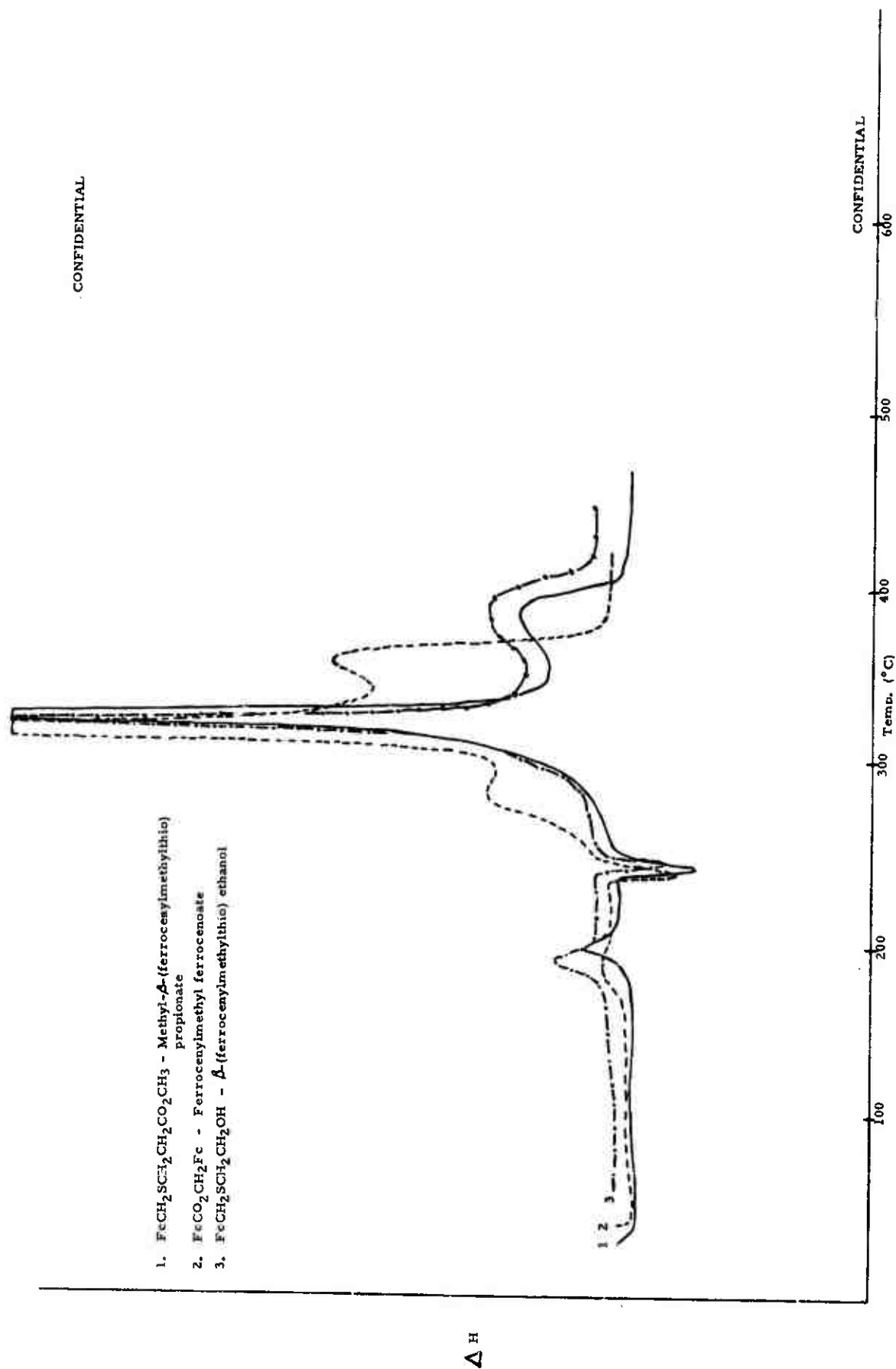


Figure 14. Differential Thermal Analyses of methyl- β -(ferrocenylmethylthio) propionate, ferrocenylmethyl ferrocenoate, and β -(ferrocenylmethylthio) ethanol.

CONFIDENTIAL

CONFIDENTIAL

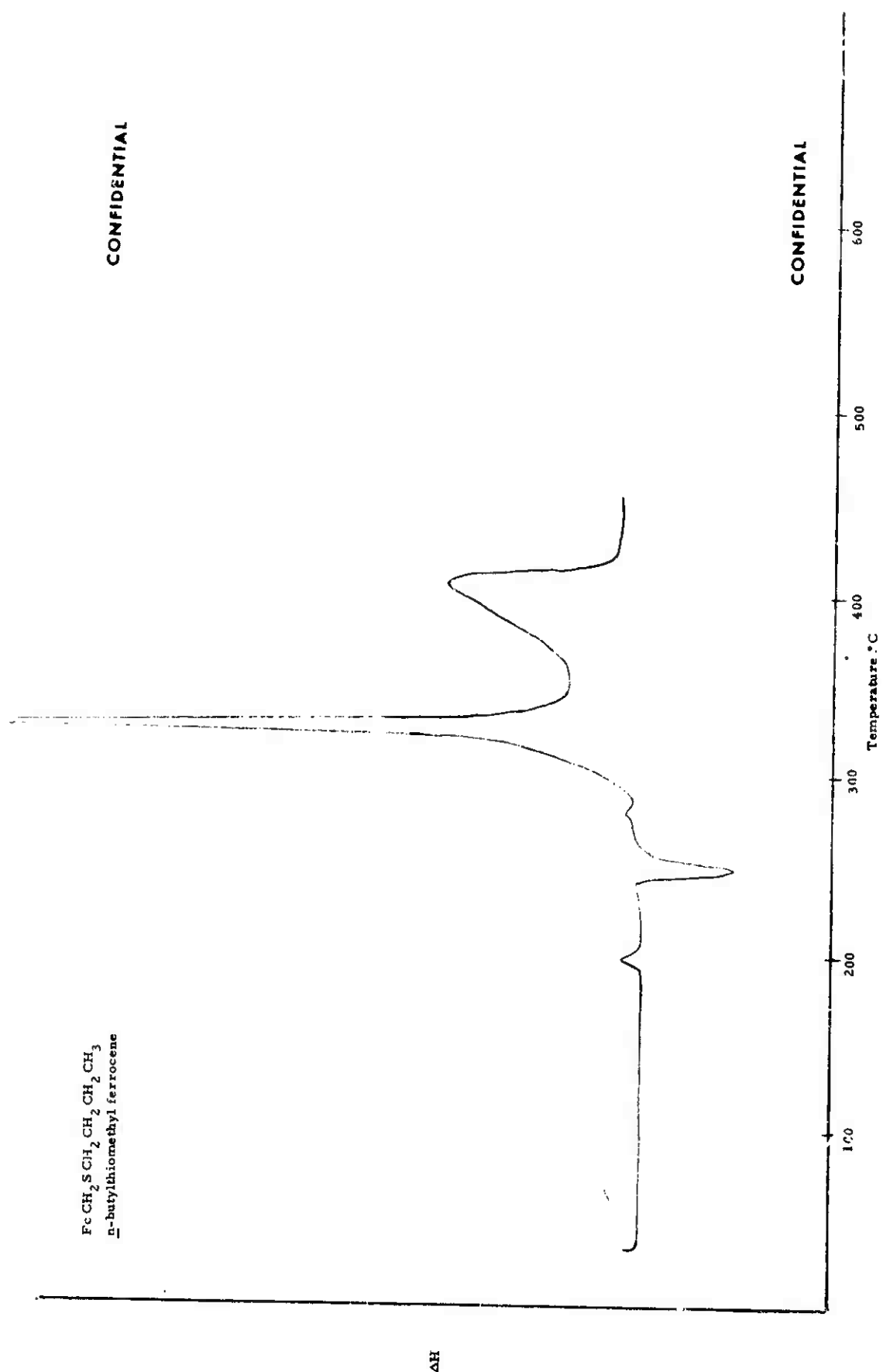


Figure 15. Differential Thermal Analysis of Ammonium Perchlorate (ung) + 5% Catalyst.

CONFIDENTIAL

CONFIDENTIAL

CONFIDENTIAL

Differential Thermal Analysis
of 5% Catalyst with AP

1. $\text{FcCH}_2\text{COCH}_2\text{CH}_3$ Ethyl Ferrocenylacetate ———
2. $\text{FcCH}_2\text{COCH}_3$ Methyl Ferrocenylacetate - - - - -
3. $\text{FcCH}_2\text{COCH}_2\text{CH}_3$ Ethyl Ferrocenylacetate - - - - -
4. $\text{FcCH}_2\text{COCH}_2\text{CH}_2\text{Fc}$ Bis(1-ferrocenyl) ether - - - - -
5. $\text{FcCH}_2\text{COCH}_2\text{CH}_3$ Methyl Ferrocenylacetate - - - - -
6. $\text{FcCH}_2\text{COCH}_2\text{CH}_2\text{CH}_2\text{Fc}$ Bis(1-ferrocenyl) ether - - - - -

ΔH

CONFIDENTIAL

CONFIDENTIAL



Figure 16. DTA of 5 Percent Catalyst with AP

CONFIDENTIAL

(U) Figures 17, 18 and 19, show percent weight loss by TGA method of:

- a. Copper (II) complex of β -ferrocenyl- β -oxo-propionaldehyde.
- b. Dimethylaminomethyl ferrocene.
- c. 1,3-diferrocenyl-1 oxo-2-propene.

(C) Visual observations show that ethyl ferrocenylacetate (which contains one percent *n*-butylhydroquinone) oxidizes rapidly at room temperature when exposed to the atmosphere and oxidizes slightly when sealed. Also, *n*-butyl thiomethyl ferrocene shows slight oxidation at room temperature after several weeks.

(U) b. Burn Rate Studies

(C) A 1000 gram batch of HC propellant without burning rate catalyst and curing agent was made and 100 gram batches of propellant containing rate catalyst and curing agent were prepared from the master batch. Composition of the propellant used is shown below:

CONFIDENTIAL

<u>Composition</u>	<u>Weight, %</u>
HC Binder	13.30
MAPO	0.70
AP (unground)	27.20
AP (9600)	40.80
Al (H-5)	16.00
Catalyst	2.00

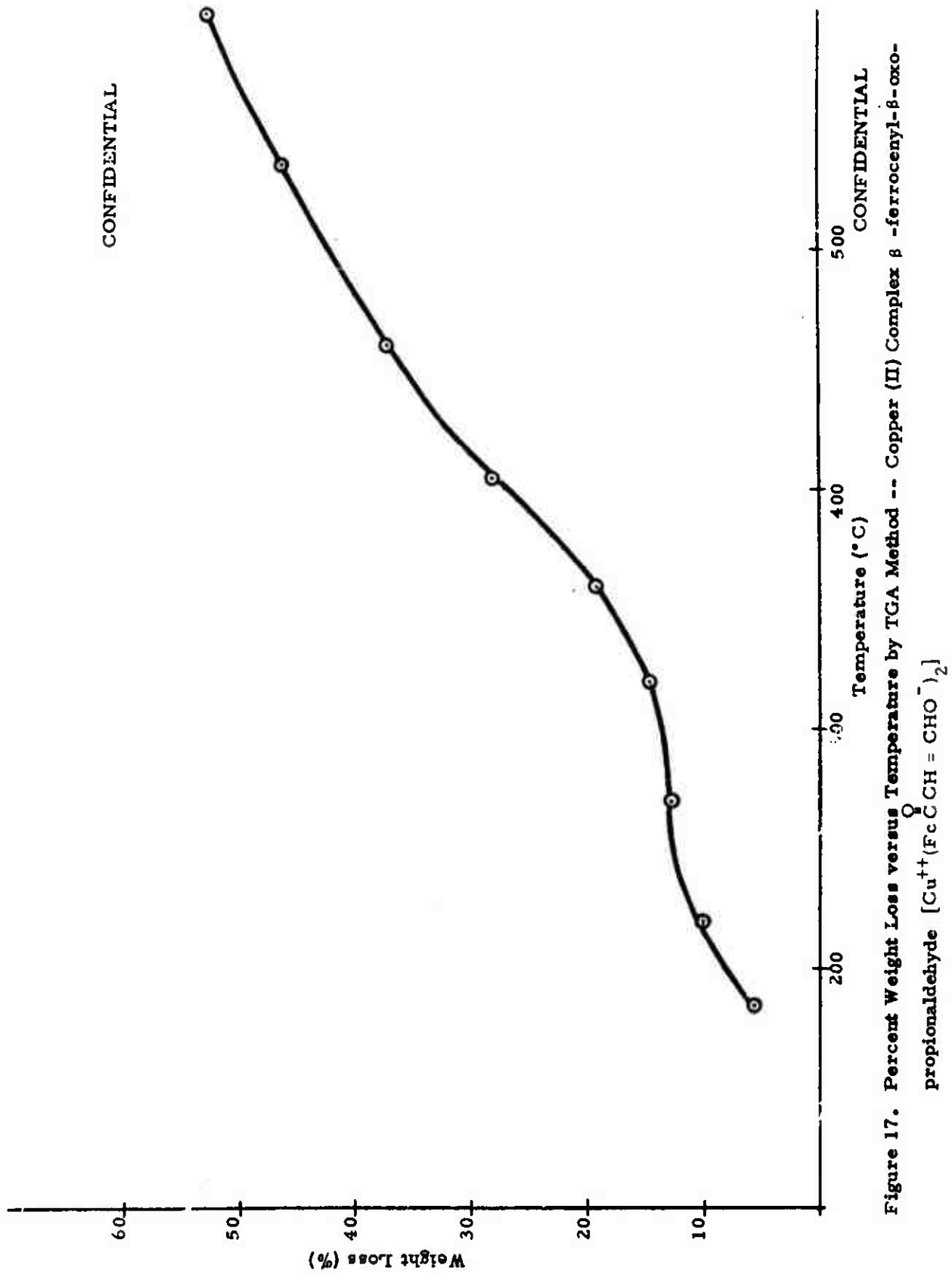
CONFIDENTIAL

(C) Propellant mixes containing methoxymethyl ferrocene and *n*-butyl ferrocene [P-IV^(R)], in concentrations of two percent, were prepared and evaluated for a comparison of burning rates. The evaluation was conducted in the pressure range of 400 - 1200 psig and the propellants were burned as strands. The data obtained (Figure 20) indicates that methoxymethyl ferrocene yields an increase in burning rate and displays a lower pressure exponent.

(C) Addition of dimethylaminomethyl ferrocene to the mix (rate catalyst added last in all cases) caused immediate release of ammonia gas, which was noted until the end of vacuum mixing (25 minutes after addition of the catalyst). Several explanations can be offered as to why ammonia gas evolved:

CONFIDENTIAL

CONFIDENTIAL



CONFIDENTIAL

CONFIDENTIAL

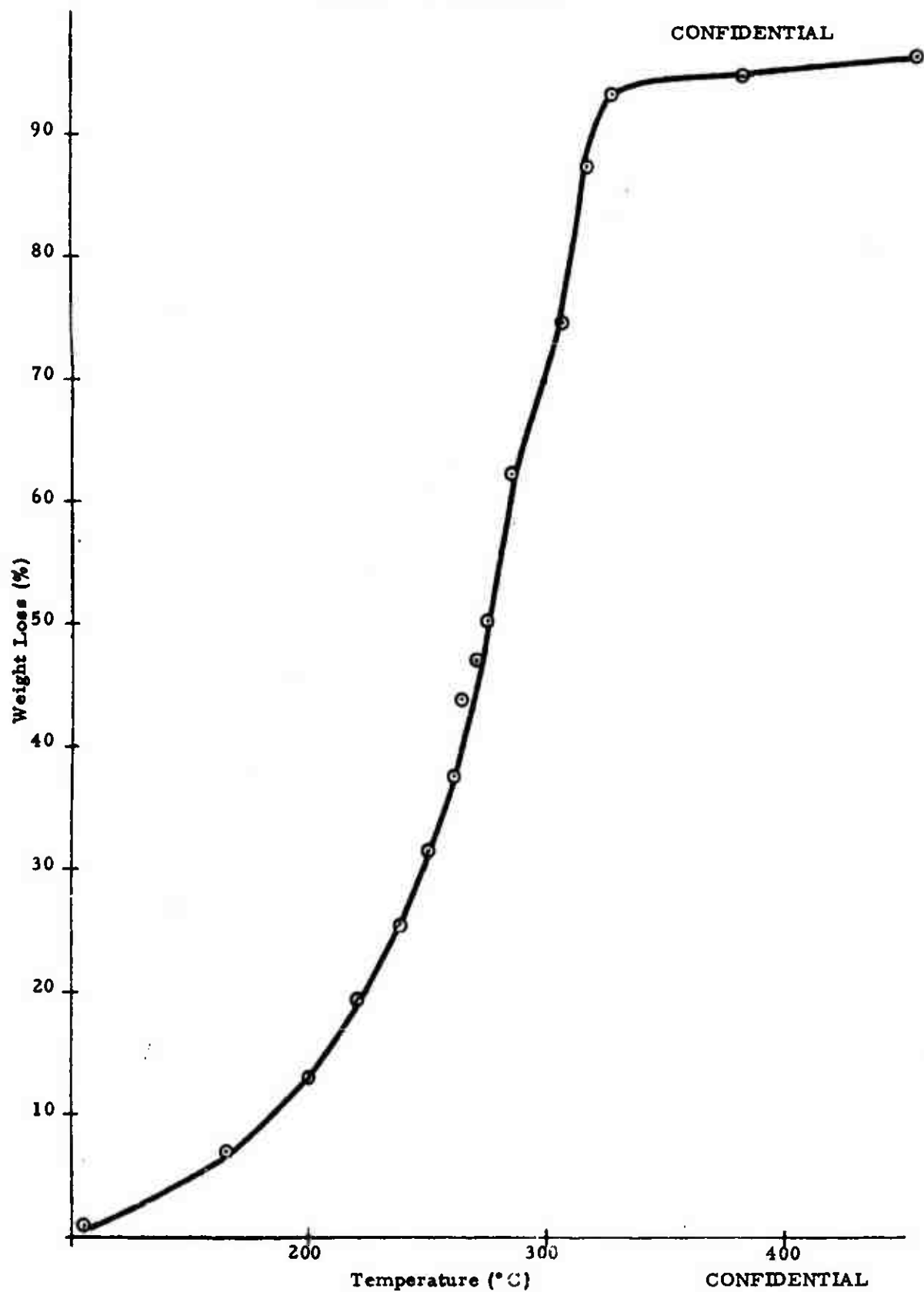


Figure 18. Percent Weight Loss versus Temperature by TGA Method -
dimethylaminomethyl ferrocene $[\text{FcCH}_2\text{N}(\text{CH}_3)_2]$

CONFIDENTIAL

CONFIDENTIAL

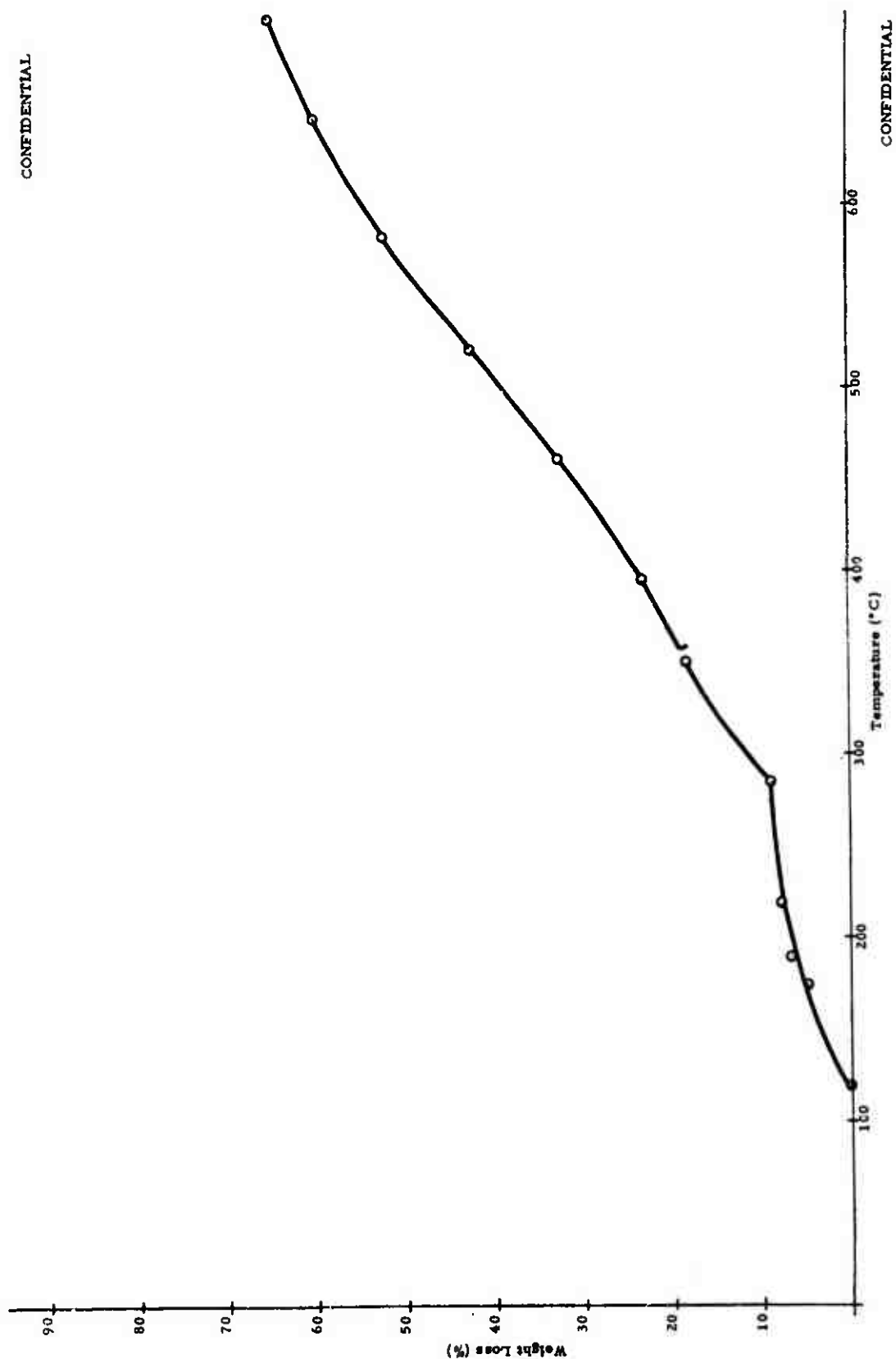


Figure 19. Percent Weight Loss versus Temperature by TGA Method --
1,3-diferrocenyl-1 oxo-2-propene ($\text{FcCH} = \text{CH} \overset{\text{O}}{\underset{\text{O}}{\text{C}}} \text{Fc}$)

CONFIDENTIAL

CONFIDENTIAL

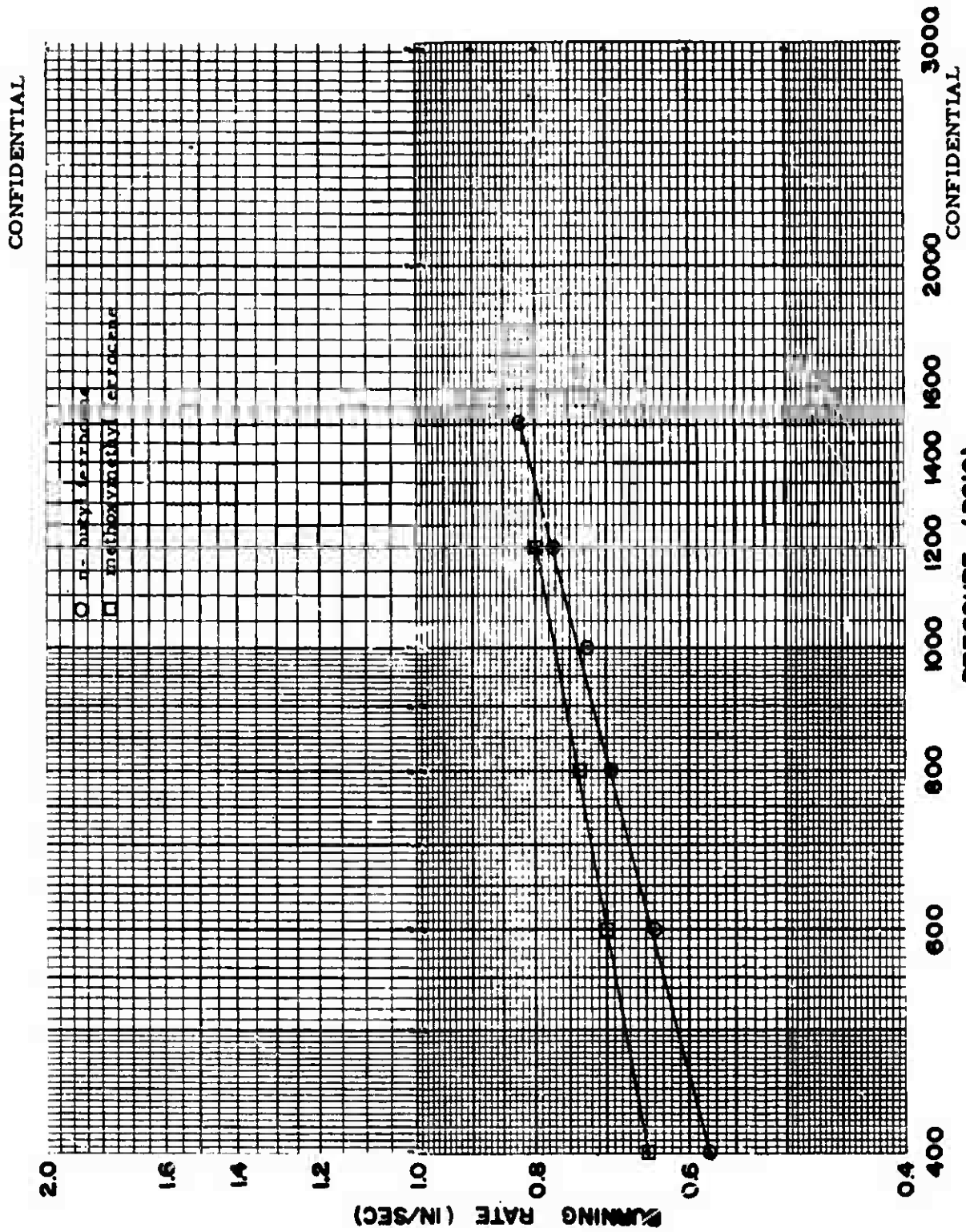


Figure 20. Comparison of the Burning Rate Characteristics of Propellants Catalyzed with PLASTISCAT-IV(R) and Methoxymethyl Ferrocene.

CONFIDENTIAL

CONFIDENTIAL

1. dimethylaminomethyl ferrocene reacted with ammonium perchlorate.
2. dimethylaminomethyl ferrocene decomposed
3. dimethylaminomethyl ferrocene reacted with MAPO

The latter explanation would appear to be the most plausible since the propellant failed to cure in 7 days at 135° F. Consequently, no data are available as yet to show the burning rate potential of this material.

(U) Other catalysts evaluated were:

1,3-diferrocenyl-1 oxo-2 propene
bis (α -ferrocenylethyl) ether
alloxymethyl ferrocene
ethoxymethyl ferrocene

(C) The burning rates of propellant containing the catalysts evaluated are illustrated on Figure 21. These data show that although 1,3-diferrocenyl-1 oxo-2 propene did not disperse well, the burning rate of propellant containing this catalyst is considerably better than that of the propellant containing n-butyl ferrocene [PLASTISCAT-IV^(R)]. Propellants containing methoxymethyl ferrocene and bis (α -ferrocenylethyl) ether also demonstrated a higher burning rate than propellant containing n-butyl ferrocene. As predicted, bis (α -ferrocenylethyl) ether, alloxymethyl ferrocene and ethoxymethyl ferrocene lowered the propellant viscosity to the 20 to 25 Kp. range. Observations concerning the mechanical properties and sensitivity of these formulations are tabulated in Table IV.

(U) The burn rate characterization of a further series of catalysts has been accomplished. The catalysts evaluated were:

methyl ferrocenylacetate
ethyl ferrocenylacetate
methyl 8 - (ferrocenylmethylthio) propionate
poly- [bis (methythiomethyl) ferrocene]

(C) Characteristics of the cured propellants are summarized in Table V. During cure, a tough, leathery "skin" formed on the surface of each of the propellants containing methyl ferrocenylacetate and ethyl ferrocenylacetate. A skin was also observed on the surface of samples of these propellants, which had been removed from the mixer and allowed to cool to ambient temperature. In the latter case, the skin was not as thick or tough as that on the cured propellant.

CONFIDENTIAL

TABLE IV

PHYSICAL PROPERTY CHARACTERISTICS AND SENSITIVITY TEST RESULTS OF CATALYZED PROPELLANT

Catalyst	SHORE "A" DURO	Cure Time (hrs)	Physical Property	Friction Sensitivity (rpm)	Impact Sensitivity (E ₀)
1,3-diferrocenyl-1 oxo-2 propene	50	48	good	5000 +	54
bis (α -ferrocenylethyl) ether	45	48	slightly plastic	5000 +	155
alloxymethyl ferrocene	45	48	slightly plastic	5000 +	150
ethoxymethyl ferrocene	55	72	high modulus	5000 +	125
dimethylaminomethyl ferrocene	Did not cure			5000 +	98

CONFIDENTIAL

CONFIDENTIAL

CONFIDENTIAL

CONFIDENTIAL

TABLE V
CHARACTERISTICS OF CATALYZED PROPELLANT

<u>Catalyst</u>	<u>Shore "A" DURO</u>	<u>Cure Time</u>	<u>Physical Properties</u>
Methyl ferrocenylacetate	52	44	Good. Forms tough "skin" on top during cure.
Ethyl ferrocenylacetate	46	68	Low modulus, slightly plastic.
Methyl β -(ferrocenylmethylthio) propionate	34	68	Very low modulus.
Poly-[bis(methylthiomethyl) ferrocene]	53	40	Good.

CONFIDENTIAL

CONFIDENTIAL

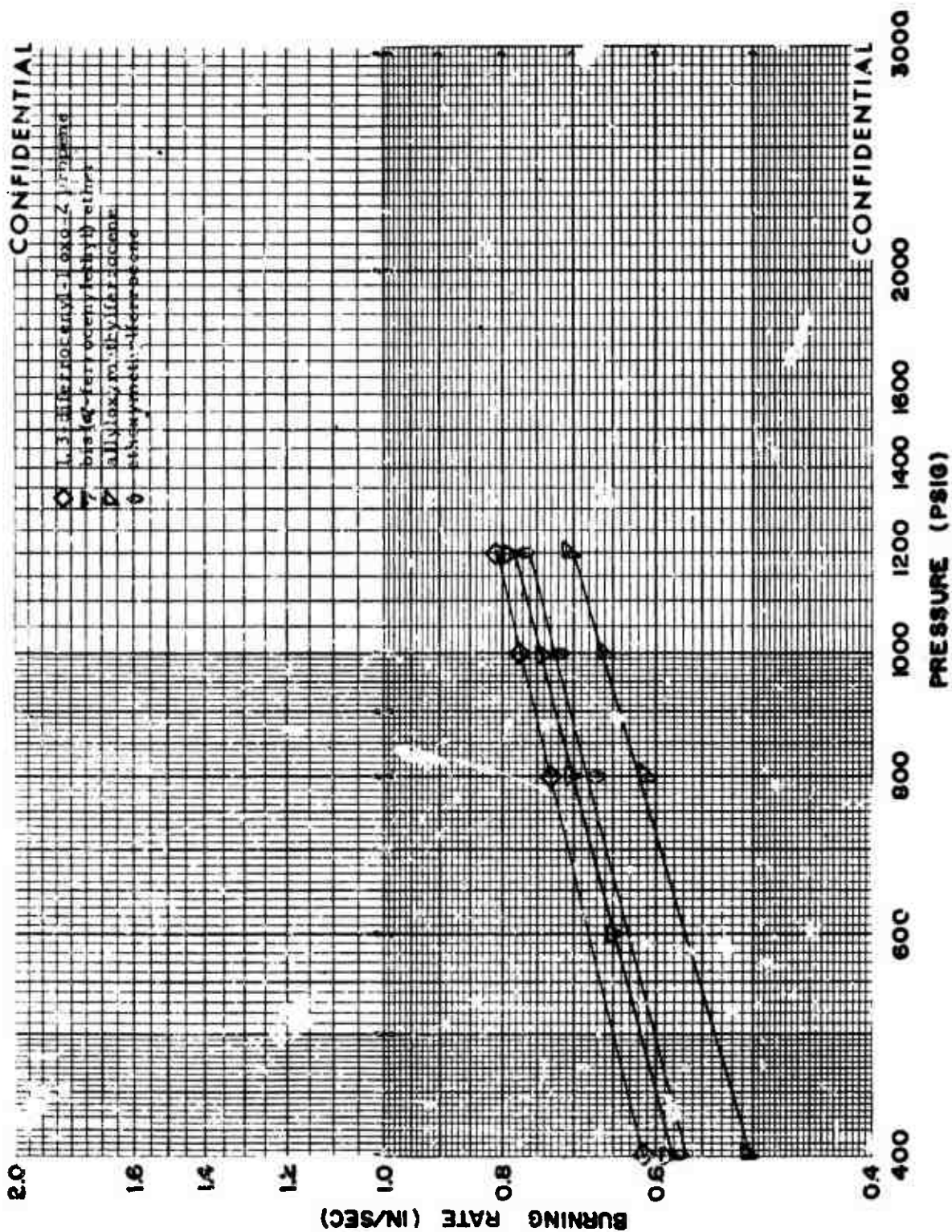


Figure 21. Comparison of Burning Rate Characteristics of Propellants Catalyzed with 1,3-diferrocenyl-1-oxo-2-propene, bis(4-ferrocenylethyl) ether, allyloxymethyl ferrocene and ethoxymethyl ferrocene.

CONFIDENTIAL

(C) Addition of methyl ferrocenylacetate and ethyl ferrocenylacetate to the mixer caused a significant drop in viscosity of each propellant to the 15 to 20 kilopoise range. Addition of methyl β -(ferrocenylmethylthio) propionate to the propellant mix effected a viscosity reduction to the 20 to 25 kilopoise range. No reduction in viscosity was experienced with the addition of mercaptomethyl terminated poly-[bis (methylthiomethyl) ferrocene].

(C) A comparison of the characteristics of the catalyzed propellants can be made with that of n-butyl ferrocene on Figures 20, 22 and 23. These data show the burning rate characteristics for ethyl ferrocenylacetate to be slightly higher than that of n-butyl ferrocene, while the burning rate of methyl ferrocenylacetate is equal to that of n-butyl ferrocene. A propellant mix containing copper (II) complex of β -ferrocenyl- β -oxo-propionaldehyde was still in cure at the end of this report period. Data from this mix will be reported in a subsequent report.

CONFIDENTIAL

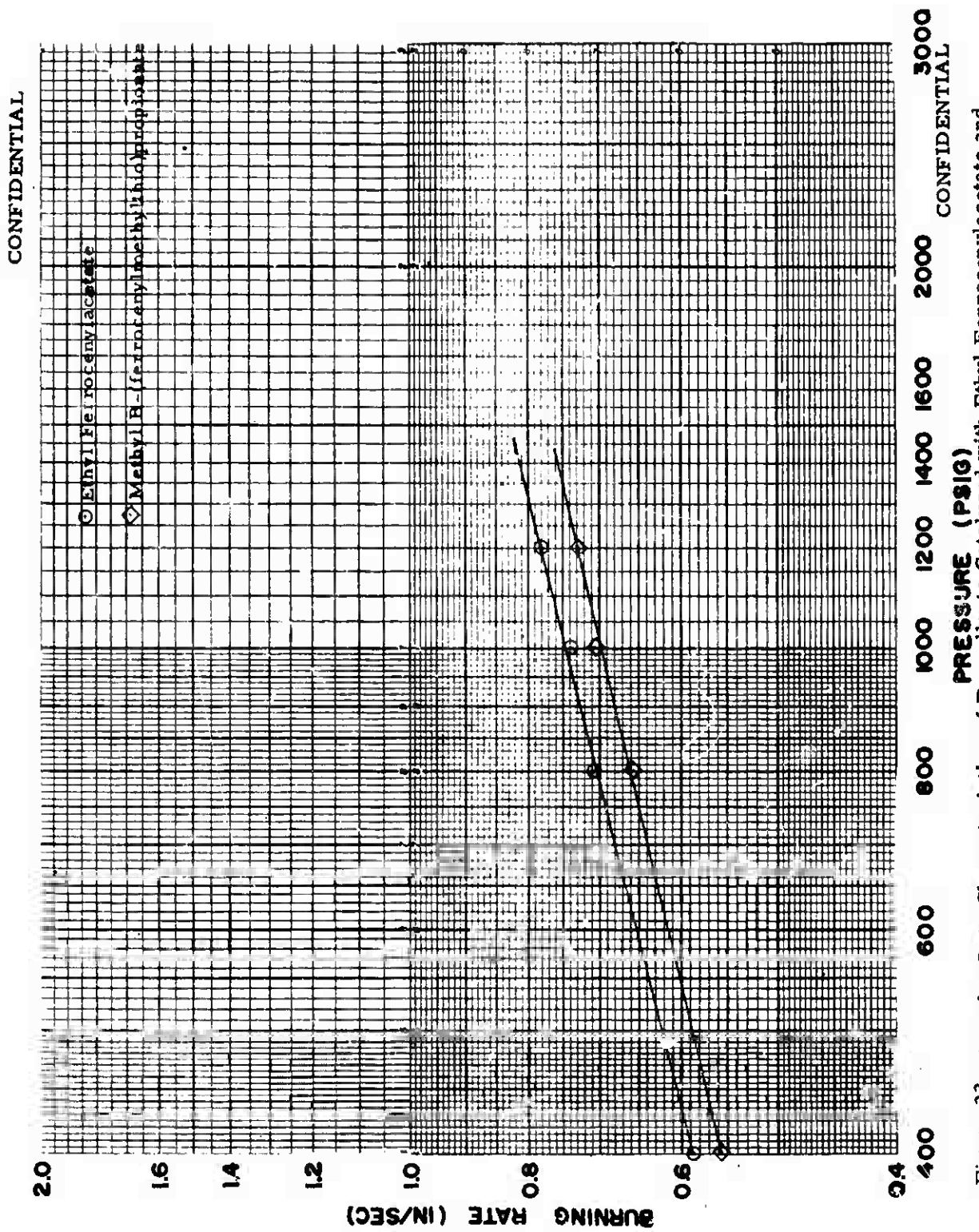


Figure 22. Burning Rate Characteristics of Propellants Catalyzed with Ethyl Ferrocenylacetate and Methyl-B-(ferrocenylmethyl)propionate.

CONFIDENTIAL

CONFIDENTIAL

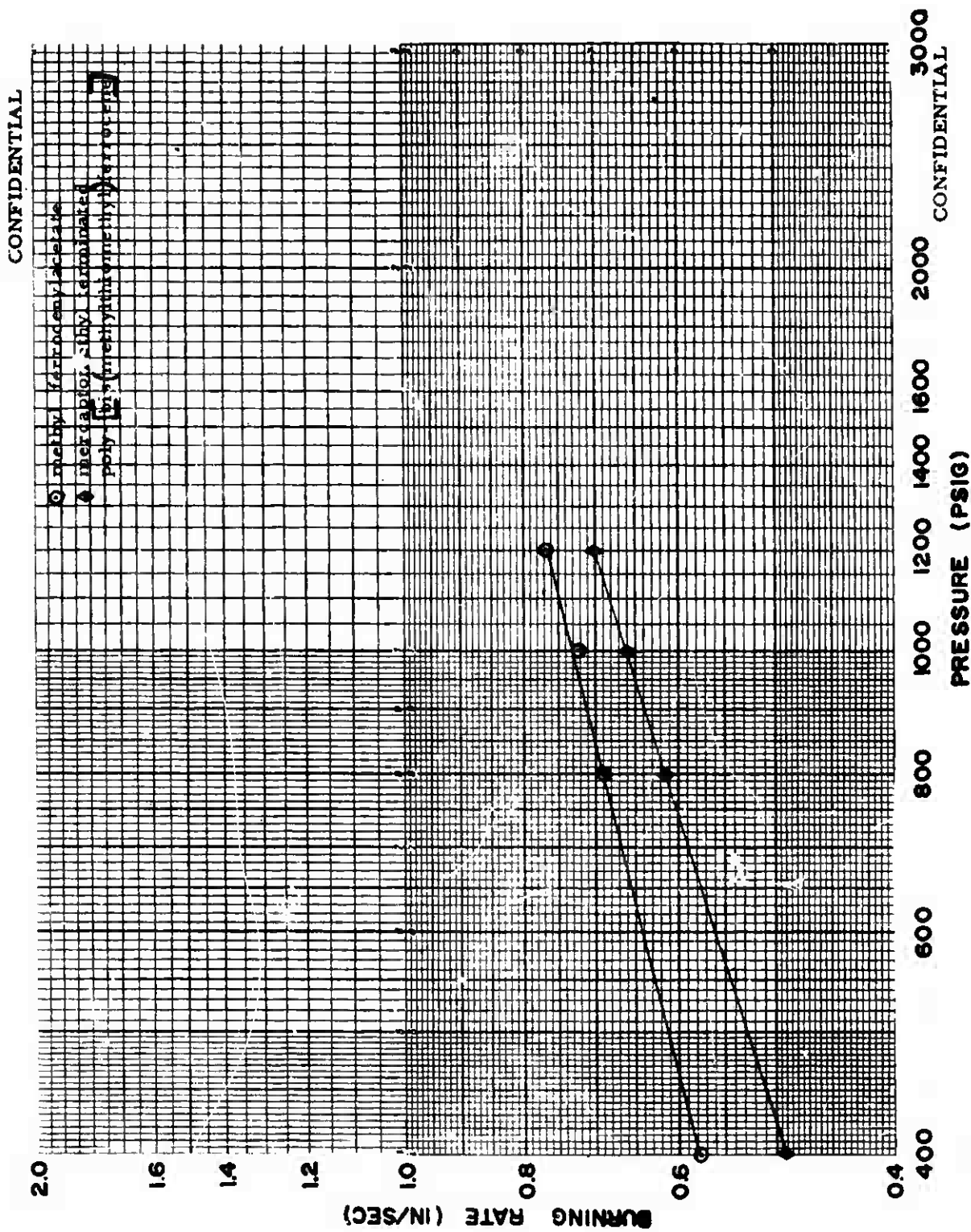


Figure 23. Burning Rate Characteristics of Propellants Catalyzed with Methyl Ferrocenylacetate and Mercaptoethyl Terminated poly-[bis(methylthiomethyl)ferrocene].

CONFIDENTIAL

CONFIDENTIAL

SECTION III

CONCLUSIONS

1. (U) Thirty candidate catalyst materials have been synthesized and the physical properties ascertained.
2. (C) The previously reported freezing points of the candidate catalysts have been found to be in error due to extreme degrees of supercooling which seem to be characteristic of liquid ferrocene derivatives. Melting points, which are not as time dependent as freezing points, will be reported in the future.
3. (U) Compatibility testing of the catalyst materials with oxidizer, curing agents, etc., are essentially complete and, with the exception of dimethylaminomethyl ferrocene, no serious problems are expected in the formulation and scale-up of propellants containing the candidate catalysts.
4. (C) Burn rate studies thus far indicate that several prospective liquid ferrocene derivatives offer considerable potential in increasing burn rates over that presently available with n-butyl ferrocene [PLASTISCAT-IV^(R)].

CONFIDENTIAL

CONFIDENTIAL

Security Classification

DOCUMENT CONTROL DATA - R&D		
(Security classification of title, body of abstract and indexing annotation must be entered when the overall report is classified)		
1. ORIGINATING ACTIVITY (Corporate author) Thiokol Chemical Corporation Huntsville Division Huntsville, Alabama	2a. REPORT SECURITY CLASSIFICATION <div style="text-align: center; font-weight: bold;">CONFIDENTIAL</div>	
	2b. GROUP <div style="text-align: center;">IV</div>	
3. REPORT TITLE Combustion Mechanism of High Burning Rate Solid Propellants		
4. DESCRIPTIVE NOTES (Type of report and inclusive dates) Quarterly Technical Report -- 1 March through 31 May 1967		
5. AUTHOR(S) (Last name, first name, initial) Flanigan, David A.		
6. REPORT DATE June 1967	7a. TOTAL NO. OF PAGES i - v, 1 - 43	7b. NO. OF REFS ---
8a. CONTRACT OR GRANT NO. F04611-67-C-0034 b. PROJECT NO. 3148 c. d.	9a. ORIGINATOR'S REPORT NUMBER(S) <div style="text-align: center;">AFRPL-TR-67-165</div>	
	9b. OTHER REPORT NO(S) (Any other numbers that may be assigned this report) 39-67 (Control No. C-67-39A)	
10. AVAILABILITY/LIMITATION NOTICES In addition to security requirements which must be met this document is subject to special export controls and each transmittal to foreign governments or foreign nationals may be made only with prior approval of AFRPL (RPPR-STINFO), Edwards, California 93523		
11. SUPPLEMENTARY NOTES <div style="text-align: center;">---</div>	12. SPONSORING MILITARY ACTIVITY Air Force Rocket Propulsion Laboratory Research and Technology Division Edwards Air Force Base, California 93523	
13. ABSTRACT (C) Thiokol's program to tailor the burning rate of a propellant predictably and controllably to any desired level in the range from one to ten inches per second is divided into three phases: Phase I-Synthesis of Burning Rate Catalysts, Phase II-Decomposition Studies and Evaluation of Catalysts and Phase III-Decomposition of Advanced Oxidizers, Fuels, and Binders. Experiments will be performed to synthesize more efficient burn rate catalysts by maximizing already determined guideline properties toward development of an ideal catalyst under Phases I and II. Following catalyst synthesis, each compound showing potential will be subjected to comprehensive decomposition studies and combustion mechanism evaluation with propellant ingredients. To date, 30 candidate catalyst materials have been synthesized and the physical properties ascertained. Compatibility testing of these materials is essentially complete and no serious problems are expected in the formulation and scale-up of propellants containing the candidate catalysts. Burn rate studies thus far indicate that several prospective liquid ferrocene derivatives offer considerable potential in increasing burn rates over that presently available with n-butyl ferrocene [PLASTISCAT-IV ^(R)].		

DD FORM 1 JAN 64 1473

CONFIDENTIAL
Security Classification

CONFIDENTIAL

CONFIDENTIAL
(This page is UNCLASSIFIED)

CONFIDENTIAL

Security Classification

14 KEY WORDS	LINK A		LINK B		LINK C	
	ROLE	WT	ROLE	WT	ROLE	WT
Combustion mechanism Burn rate Catalyst synthesis Decomposition studies						

INSTRUCTIONS

1. ORIGINATING ACTIVITY: Enter the name and address of the contractor, subcontractor, grantee, Department of Defense activity or other organization (corporate author) issuing the report.

2a. REPORT SECURITY CLASSIFICATION: Enter the overall security classification of the report. Indicate whether "Restricted Data" is included. Marking is to be in accordance with appropriate security regulations.

2b. GROUP: Automatic downgrading is specified in DoD Directive 5200.10 and Armed Forces Industrial Manual. Enter the group number. Also, when applicable, show that optional markings have been used for Group 3 and Group 4 as authorized.

3. REPORT TITLE: Enter the complete report title in all capital letters. Titles in all cases should be unclassified. If a meaningful title cannot be selected without classification, show title classification in all capitals in parenthesis immediately following the title.

4. DESCRIPTIVE NOTES: If appropriate, enter the type of report, e.g., interim, progress, summary, annual, or final. Give the inclusive dates when a specific reporting period is covered.

5. AUTHOR(S): Enter the name(s) of author(s) as shown on or in the report. Enter last name, first name, middle initial. If military, show rank and branch of service. The name of the principal author is an absolute minimum requirement.

6. REPORT DATE: Enter the date of the report as day, month, year, or month, year. If more than one date appears on the report, use date of publication.

7a. TOTAL NUMBER OF PAGES: The total page count should follow normal pagination procedures, i.e., enter the number of pages containing information.

7b. NUMBER OF REFERENCES: Enter the total number of references cited in the report.

8a. CONTRACT OR GRANT NUMBER: If appropriate, enter the applicable number of the contract or grant under which the report was written.

8b, 8c, & 8d. PROJECT NUMBER: Enter the appropriate military department identification, such as project number, subproject number, system numbers, task number, etc.

9a. ORIGINATOR'S REPORT NUMBER(S): Enter the official report number by which the document will be identified and controlled by the originating activity. This number must be unique to this report.

9b. OTHER REPORT NUMBER(S): If the report has been assigned any other report numbers (either by the originator or by the sponsor), also enter this number(s).

10. AVAILABILITY/LIMITATION NOTICES: Enter any limitations on further dissemination of the report, other than those imposed by security classification, using standard statements such as:

- (1) "Qualified requesters may obtain copies of this report from DDC."
- (2) "Foreign announcement and dissemination of this report by DDC is not authorized."
- (3) "U. S. Government agencies may obtain copies of this report directly from DDC. Other qualified DDC users shall request through _____."
- (4) "U. S. military agencies may obtain copies of this report directly from DDC. Other qualified users shall request through _____."
- (5) "All distribution of this report is controlled. Qualified DDC users shall request through _____."

If the report has been furnished to the Office of Technical Services, Department of Commerce, for sale to the public, indicate this fact and enter the price, if known.

11. SUPPLEMENTARY NOTES: Use for additional explanatory notes.

12. SPONSORING MILITARY ACTIVITY: Enter the name of the departmental project office or laboratory sponsoring (paying for) the research and development. Include address.

13. ABSTRACT: Enter an abstract giving a brief and factual summary of the document indicative of the report, even though it may also appear elsewhere in the body of the technical report. If additional space is required, a continuation sheet shall be attached.

It is highly desirable that the abstract of classified reports be unclassified. Each paragraph of the abstract shall end with an indication of the military security classification of the information in the paragraph, represented as (TS), (S), (C), or (U).

There is no limitation on the length of the abstract. However, the suggested length is from 150 to 225 words.

14. KEY WORDS: Key words are technically meaningful terms or short phrases that characterize a report and may be used as index entries for cataloging the report. Key words must be selected so that no security classification is required. Identifiers, such as equipment model designation, trade name, military project code name, geographic location, may be used as key words but will be followed by an indication of technical context. The assignment of links, rules, and weights is optional.

CONFIDENTIAL
Security Classification

CONFIDENTIAL